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# Investigation of the Early Stage Corrosion Characteristics of Molten (Nitrate) Salt – Oxide – Metal Interfaces, and the links to Interfacial Thermo Mechanical Stress

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# 12 Keyword:

13 Molten salt; Thermal stress; Stainless Steels; Nano-indentation; Multi-corrosion layer;

14 Spallation;

# 15 Highlight:

- Early-stage synergy of temperature and gas on overall corrosion were established.
- 17 Thermo-mechanical stress effects across metal/oxide/interfaces were
   18 characterised.
- Inner tensile stress linked to NaFeO<sub>2</sub> directly correlates with the tendency for
   spallation to occur.
- Tests in Ar was observed to increase corrosion rate compared to test in air.
- Compressive stress linked to inner oxides and metal correlates to improved
   resilience of interface.

# 24 Abstract:

This paper investigates the early-stage corrosion behaviour of 4 alloys in solar-salt under air and argon at elevated temperatures for 7 days. The multi-layered interfacial corrosion oxides were observed to consist of an inner Cr-rich oxide and an outer Na/Fe-rich oxide. The inner tensile stress between the interfacial oxide layers were observed to correlate to the spallation of the porous outer Na/Fe-rich oxide layer. The combination of chemical aggression of molten salt, high temperature and argon atmospheres, acts in synergy to increase the kinetics of 31 formation of Na/Fe-rich outer oxide layer, and it's tendency to degrade by spallation due to an

32 inner tensile stress.

## **33 1.Introduction**

34 During the last few decades, efforts have been made to mitigate the progress of climate change 35 through reduction of CO<sub>2</sub> emissions, and transitioning from the use of conventional coal-fired 36 and/or fossil fuel power plants to renewable energy power [1]. The exploitation of solar energy 37 has gained much more interests than other kinds of renewable energy sources due to the 38 availability, accessibility and inexhaustibility of solar thermal irradiation. This has helped to 39 significantly promote the development of the solar thermal energy conversion systems. 40 Concentrated solar power (CSP) plants, accompanied with thermal energy storage (TES) 41 systems, are the most promising and most widely used energy conversion plants, with the 42 potential to meet a variety of energy intensive industrial needs. It works by transferring the 43 abundant but intermittent solar thermal energy to continuous, dispatch-able and stable solar 44 thermal electricity [2, 3]. Due to its high potential efficiency, low operation cost and low 45 environmental impact [4], solar power collector technology is the foundation of commercial 46 CSP systems where molten salts are widely used as heat transfer fluids (HTFs) and thermal 47 storage material, and usually in direct interaction with metallic materials used as heat collectors, 48 heat exchangers/transfer pipes, and hot storage tanks [3, 5, 6]. Nitrate salts (60 wt % NaNO<sub>3</sub> 49 and 40 wt % KNO<sub>3</sub> also known as 'Solar salt'), is widely employed in CSP plants due to its 50 good thermal-physical properties; such as high heat capacity, high thermal conductivity, wide 51 working temperature range, high viscosity and low cost [6-8].

However, the low thermal stability of nitrate salt mixtures restricts the maximum operation temperature in CSPs, and hence restricts its efficiency. Rapid decomposition of nitrate ions into nitrite ions and further into soluble oxides and gaseous nitrous oxides occurs by Equation 1 and Equation 2 [9, 10] at temperature exceeding 600°C. This has already been confirmed using XRD analysis of salts before and after experiments in a previous publication [11].

$$NO_3^- \rightleftharpoons NO_2^- + \frac{1}{2}O_2(g)$$
 Equation 1

$$2NO_2^- \rightleftharpoons NO_2(g) + NO(g) + O^{2-}$$
 Equation 2

57 The chemical compositions of the salts changed due to the reactions above and thus altered the 58 thermo-properties of molten salt. Worse still, this increases the oxidising and corrosive 59 tendencies of the molten salt species to increase rate of degradation of structural metallic 60 materials used in CSP systems [12-15], particularly at temperatures above its thermal stability 61 temperatures. The effect of temperature on the corrosive tendencies of molten salts has also 62 recently gained increased interest with the push towards the super-critical CO<sub>2</sub> (Brayton) power 63 cycles that would require outlet temperatures from solar receiver towers in excess of 700 °C. 64 This is aimed at increasing efficiency of next generation CSP systems and reducing the 65 levelized cost of electricity (LCOE) [16]. However, this can only be delivered with more 66 aggressive salt types such as chloride and carbonate salts [17].

67 Several other research have been summarised by Walczak.et.al [8] on the corrosion 68 performance of carbon/low-alloys steel, stainless steels and Ni-based alloys with different salts 69 at different temperatures. It was shown that the corrosion rate and the thickness of corrosion 70 oxide layer increase with increase of temperature [8]. Alan [18] investigated the corrosion of 71 alloys at 400, 500 and 680°C in Solar salt and found significant alloy surface degradation 72 occurring at a corrosion rate above 450 µm/year at 680°C, almost 10 times higher than that at 73 500°C. Pieces of research shown in Figure 1, was explored to mitigate the corrosion process 74 by controlling the gas atmosphere [19, 20]. Vignarooban [20] found that corrosion rate of C-75 276 after 42 day-immersion test in open container at 500°C was 10 times higher than in sealed 76 container at 800°C. On the other hand, S.Bell et.al [21] reported that the corrosion was even 77 more severe for C-276 tested with chloride/sulphate eutectic salt in argon than that in air. 78 Therefore, they concluded that the gas atmosphere plays a significant role in directing the 79 corrosion characteristics of a molten-salt - alloy interface. In terms of designing tests to 80 simulate conditions as closely as in real CSP systems, longer duration tests in excess of 1000 81 h are usually favoured [8, 22, 23]. However, a 7-day duration has been chosen for this study to 82 represent the early stage of the evolution of the molten salt – metal corrosion interface. The 83 purpose and focus of this study on investigating the early stages of corrosion and interfacial 84 oxide formation during molten salt – metal interactions is hinged on the fact that at this early 85 stages, the corrosion process is usually initial impurity-driven, and always results in higher 86 corrosion rate [16]. In a previous study [11], it was also confirmed that the stainless steels and 87 Ni-based alloys corrosion exhibited severe corrosion with higher corrosion rate and thicker 88 corrosion scale at 600°C in argon. That's the reason why 7 day was chosen for this study.

Some recent literatures on oxidation of alloys [24, 25] suggested that mechanical properties of
 the surface oxide layers were also critical in determining its capacity to protect the substrate

91 against further oxidation. This was on the premise that the main factor influencing the oxidation 92 resistance of the alloys is the mechanical integrity of the surface oxide layer [26]. This could 93 be linked to internally built - up of stresses [27] that arises naturally during the oxidation 94 process; due to the difference in specific volume of oxide and substrate metal, and on thermal 95 stresses that emerges during cooling; due to different thermal expansion/contraction 96 coefficients between surface oxide layers and substrate [28].



Figure 1 Nitrate and nitrite content in Solar Salt stored in open atmosphere (open symbols)
and closed atmosphere (closed symbols) at 550°C, 580°C and 600°C. Adapted from Ref [19]

97

100 Therefore, in order to fully understand the corrosion behaviour of molten salt - metal interface 101 systems, the adherence, protectiveness and nano-mechanical properties of the interfacial oxide 102 layers need to be carefully investigated and characterised. This can be correlated to the 103 tendency for material degradation by spallation, manifested as loss of corrosion resilience at molten salt – metal interfaces. This study experimentally investigated the corrosion behaviours 104 105 of two stainless steel and two Ni-based alloys at 565°C and 600°C under air and argon 106 atmospheres using a combination of gravimetric (weight loss) analysis, advanced surface and 107 subsurface characterisation techniques; including nano-mechanical characterisation. This study 108 investigates the formation and protectiveness of corrosion oxide layers in the context of the 109 synergistic effect of temperature and gas atmospheres on the overall corrosion process based on a total test duration of 7 days (168 h). A key part of this study is the investigation of the 110 111 interfacial stress profile across the molten salt – oxide – metal interface. The results from this 112 aspect of the study will help in correlating the thermal stress profile across the interface with 113 the rate and tendency of oxides spallation during molten salt induced corrosion processes.

- 114 Detailed microstructural characterizations and other methods deployed in this study are 115 provided in the subsequent sessions in this paper.

# **116 2.Experimental Procedure**

#### 117 **2.1 Material and methods**

134

118 Stainless steel (AISI 321 and 347) and Ni-based alloys (Alloy 625 and Alloy 825) are used in 119 the following experiments and the compositions are provided in Table 1. The samples were cut 120 from corresponding seamless pipes supplied by Zhejiang JIULI Hi-tech Metals Co., Ltd with 121 a dimension of 25 mm x 13 mm and 3 mm thick. The samples were prepared by grinding with 122 silicon carbide paper up to P1200 (600 grit, Ra: 0.11 µm) and polishing with 9 µm, 6 µm, 1 µm 123 and 0.25 µm diamond suspension to a mirror-like surface, rinsed with acetone and deionized 124 (DI) water, then dried using hot-air. The prepared samples were weighed on an analytical 125 balance with an accuracy of 0.01mg and then kept in a desiccator for a short time before the 126 test to prevent any pre-experiment oxidation.

Solar salt (60 wt. % NaNO<sub>3</sub> and 40 wt. % KNO<sub>3</sub>) from Alfa Aesar was used for this study with analytical reagent purity (the detailed components and impurities of solar salt is provided in the supplementary file). The salts were weighed to achieve the required ratio, mixed to achieve homogeneity, and put into four crucibles to dry at 180°C for 24 h in an oven (gas-controlled furnace for test under argon atmosphere) [29] before the commencement of experiments. According to British Standard ISO 17245:2015, the total amount of solar salt for each crucible is 413 g to meet the required volume / surface area ratio (20 ml per 1 cm<sup>2</sup> of samples' area).

	10	ole 1: 1	аноу с	compos	uion oj	AISI	521 an	ia 347,	IN 023	o ana 1	n 823 (	WI. %0).
-	Alloy	С	Mn	Si	Р	Cu	Cr	Ni	Ti	Nb	Mo	F35
	AISI 321	0.018	1.44	0.519	0.035	-	17.33	9.24	0.225	-	-	Bal.
	AISI 347	0.049	0.95	0.402	0.026	0.11	17.35	9.65	-	0.656	-	Bal.
	IN 625	0.018	0.04	0.142	0.001	0.03	21.69	62.36	0.211	3.31	8.65	3.06
	In 825	0.008	0.67	0.258	0.0119	1.75	22.6	40.18	0.943	0.025	2.88	<sup>30.32</sup>

Table 1: Alloy composition of AISI 321 and 347, IN 625 and In 825 (wt. %).

Samples of each alloy retrieved from desiccator were carefully placed in the salt with the help of tweezers in four Al<sub>2</sub>O<sub>3</sub> crucibles (OD 140mm, 70mm height) to ensure consistent contact surface areas between the samples and molten salt. The crucibles (with salt and samples) were kept inside the oven for extra 12 hours to allow the required Ar gas flowing through for fully established Ar atmosphere. Then the oven was gradually heated up to test temperature (565 and 600°C for this study) and kept for 7 days (168 h) within the chamber under either air or Ar atmosphere. 6 parallel samples of each alloy are used in this study and 3 samples are washedand weighed for corrosion rate calculation and others go through microstructural analysis.

#### 147 **2.2 Corrosion Product Analysis**

148 The methodology used in calculating the mass loss and corrosion rate of the samples after the 149 test has already been defined in previous study [11] and provided in supplementary files. The 150 top-view and cross-sectional morphology of the interfacial corrosion oxides were observed 151 using Scanning Electron Microscope (SEM) with Energy Dispersive Spectroscopy (EDS). The chemistries of interfacial corrosion oxides were assessed using a combination of X-Ray 152 153 Diffraction (XRD) spectrum and Raman spectroscopy. The XRD penetration depth are around 154 19.9 µm for stainless steel and 17.6 µm for Ni-based alloys at 90° incident angle (from software 155 'Highscore'). For Raman spectroscopy, the Renishaw InVia spectrometer (UK) was used in 156 this study. 488 nm wavelength laser operated at a laser power of 0.1 and 0.5 mW with 50x 157 short distance objectives.

158 The nano-mechanical properties of interfacial corrosion oxide layers and base material were 159 analysed after nano-indentation test. The tests were performed using a Micro-materials 160 Vantage platform equipped with a Berkovich tip. Indentations were made at more than 15 161 preselected points at 20mN maximum load. Loading and unloading were at fixed time of 15 162 seconds, with a dwell time of 5 sec to allow elastic relaxation of the materials before unloading. Hardness (H) and reduced modulus (E<sub>r</sub>) were determined using the Oliver and Pharr method 163 164 [30]. The Young's (elastic) modulus of the material being measured can then be determined 165 using Equation 3,

166

$$\frac{1}{E_r} = \frac{(1-v^2)}{E} + \frac{(1-v_i^2)}{E_i}$$
 Equation 3

167

Where  $E_r$  is the reduced modulus,  $v_i$  and  $E_i$  are the Poisson's ratio and modulus of the indenter tip respectively, and v and E are the Poisson's ratio and elastic modulus of the sample. The diamond Berkovich indenter tip was assumed to have  $E_i=1141$  GPa and  $v_i=0.07$ .

#### 171 2.3 Methodology of residual stress calculation

As previously mentioned in this paper, there are potential internal stresses across the interfacial oxide layers linked to their growth mechanism, and thermal stress related to the conditions at the corroding interface. The internal stress induced by oxide growth is due to the fact that the specific volume of the oxide is rarely the same as that of the substrate metal [27], and can be mathematically related to the Pilling-Bedworth ratio (PBR) in Equation 4:

$$PBR = \frac{V_{ox}}{V_m}$$
 Equation 4

177 Where  $V_{ox}$  and  $V_m$  represent the Volume of oxide and metal. Generally, the oxide is expected 178 to be under compressive stress if PBR > 1 (the case for most metals), while the tensile stress 179 develops in oxide scale with PBR < 1 [24, 28]. Generally metal – oxide interfaces with tensile 180 stresses in the oxide cannot maintain protective films, e.g., K, Mg, Na. The PBR of NiO, Cr<sub>2</sub>O<sub>3</sub> 181 and Fe<sub>2</sub>O<sub>3</sub> are 1.65, 2.07 and 2.14 respectively [28], indicating the protective nature of these 182 oxide layer.

In addition, the thermal stress is generated during changes and fluctuations in the thermal profiles across the corroding interfaces due to the difference in the thermal expansion coefficient of the metal and oxide [28].

Note: The authors acknowledge the difficulty in isolating the stress effects due to oxide formation at high temperature and the stress contribution from the cooling of the test specimen at the end of the experiment. It is therefore assumed that both high temperature oxide formation process and cooling of test specimen contribute to the total thermal stress effects at the molten salt – oxide – metal interface, and hence, the overall spallation behaviours of the interfacial oxide layers. This has also been considered collectively in the terms of Equation 5.

192 The thermal stress caused by the thermal expansion misfit can be expressed by the Equation 5193 [28] (the derivation part is provided in supplementary files):

$$\sigma_{Ox} = \frac{-(\alpha_{Ox} - \alpha_M)\Delta T}{\frac{2t_{Ox}(1 - \gamma_M)}{t_M E_M} + \frac{(1 - \gamma_{Ox})}{E_{Ox}}}$$
Equation 5

194 Where  $\sigma$  is the stress of the oxide formed for different thermal profiles.

195  $\alpha_M$  and  $\alpha_{Ox}$  are the linear thermal-expansion coefficients for the metal and oxide (assumed 196 constant), respectively.  $\Delta T = T_L - T_H$  is the difference between the oxide formation 197 temperature (T<sub>H</sub>) and cooling temperature at the end of experiment (T<sub>L</sub>), and thus the negative 198 in sign in Equation 5; t<sub>M</sub> and t<sub>Ox</sub> are the thickness of metal and a single-oxide scale respectively. 199  $\gamma_M$  and  $\gamma_{Ox}$  is the Passion ratio of the metal and oxide, respectively.  $E_M$  and  $E_{Ox}$  are the elastic 190 modulus of the metal and oxide, respectively. 201 Referring to Equation 5, the  $\Delta T$  is assumed negative and  $\alpha_{Ox}$  is typically  $< \alpha_M$ , and the 202 resultant stress (negative sign) in the oxide is typically compressive.

## 203 **3. Results**

**3.1 The comparisons of corrosion rate measurements** 

Figure 2 shows the measured weight loss and corrosion rate of stainless steel and Ni-based alloys after 7 – day immersion test at 565 and 600°C under air and argon atmospheres. The results showed an expected trend of higher corrosion rate of stainless steels than Ni-based alloys in molten salt conditions, as well as an expected general increase in corrosion rate with increase in temperature from 565 to 600°C in air. This along with the error bars from Figure 2 repeated experiments helps to demonstrate the reliability and repeatability of the experiments.

211 It is also important to note that the highest corrosion rate was recorded after immersion test at 212 600°C under Ar atmosphere for all alloys; ~ twice as high as the corrosion rate of test under air 213 atmosphere for AISI 321, 347 and In 825, and ~ 10 times higher for IN 625. For IN 625, the 214 corrosion rate increases from 11.9  $\mu$ m/y in the air to 128  $\mu$ m/y under Ar atmosphere at 600°C. 215 Although IN 625 has been reported to have a superior corrosion resistance compared to 216 stainless steels [18, 31] in air below 600°C with corrosion rate of 12.7 µm/y, however, such 217 superiority in corrosion resistance reduces at temperatures in excess of 680°C in air with 218 corrosion rate of 594 µm/y. For AISI 321, AISI 347 and In 825, the corrosion rate was observed 219 to increase steadily with temperature and with change of gas atmosphere from air to Ar at 220 600°C in Figure 2 after 7 days.



Figure 2 Mass loss (and corrosion rate) of AISI 321 AISI 347 IN 625 and In 825 specimens at different temperature and atmosphere in solar salt for 7 days

#### 224 **3.2** Corrosion Oxide Characterisation – XRD Analysis

225 According to the XRD analysis, the chemistry of corrosion oxides formed on the surfaces of 226 both stainless samples; AISI 321 and 347 in air are similar, with the only difference being in terms of the intensity of the identified reflections. Based on the XRD results shown in Figure 227 228 3 (a1) (a2) (b1) (b2), it can be concluded that the interfacial corrosion oxide layers consist of 229 iron oxide ( $Fe_2O_3$ ), iron chromium spinel ( $FeCr_2O_4$ ), sodium ferrite (NaFeO<sub>2</sub>), iron chromium 230 oxide and chromium oxide (Cr<sub>2</sub>O<sub>3</sub>). XRD patterns show more variability in oxide chemistry at 231 600°C than at 565°C, indicating the formation of a more complex combination of oxide spinel 232 at 600°C. In contrast to the corrosion oxides formed under Ar atmosphere (Figure 3 (a3), (b3)), 233 the Cr-containing oxide was present only in form of FeCr<sub>2</sub>O<sub>4</sub> without forming Cr<sub>2</sub>O<sub>3</sub>. This is 234 related to the oxide forming corrosion reaction most thermodynamically affected by the 235 availability of oxygen/oxide ions. The corrosion oxides formed on the surface of IN 625 in air 236 is mainly NiO, while Cr<sub>2</sub>O<sub>3</sub> and FeCr<sub>2</sub>O<sub>4</sub> was identified on the surface of IN 625 samples under 237 Ar. This finding correlates well with weight loss results in Figure 2, where IN 625 samples 238 under Ar atmospheres exhibited higher corrosion rate which is directly linked to increased 239 number of interfacial corrosion oxides formed on the surface. The main corrosion oxides 240 formed on In 825 are NiO, Fe<sub>2</sub>O<sub>3</sub>, FeCr<sub>2</sub>O<sub>4</sub> and Cr<sub>2</sub>O<sub>3</sub>. NaFeO<sub>2</sub> was formed when test was 241 performed at 600°C under air and Ar atmospheres, indicating the effect of both temperature 242 and gas atmospheres on the formation kinetics of NaFeO<sub>2</sub>.

#### 243 **3.3 Corrosion Oxide Characterisation – SEM Surface Analysis**

The oxide layers formed on the surface of samples are expected to provide some level of 244 245 protection to the base metal from continuous molten salt penetration. Therefore, interfacial 246 oxide formation process and mechanisms are critical to the overall corrosion mechanism at the 247 molten salt - metal interface. The SEM images of the top surface and possible oxide phases 248 from EDS analysis are shown in Figure 4 and provided in Table 2 respectively. Referring to 249 Figure 4 (a1) (b1) (c1) and (d1) for immersion test in air at 565°C after 7 days, the corroded 250 surfaces of stainless steels appears to be smooth with only tiny and localized bulge in the 251 stainless steel samples, while the Ni-based alloy surface shows little difference from un-252 corroded samples with visible evidence of polishing marks. This indicates that the corrosion is 253 mild in air at 565°C after 7 days. Referring to Figure 4 (a2) (a3) (b2) and (b3) for the stainless254 steel samples tested at 600°C in air and under Ar atmosphere, thicker but less compact corrosion 255 oxide layers with evidence of layer spallation were observed on metal surface at 600°C than 256 that formed at 565°C. This can be confirmed from EDS analysis shown in Table 2 (spectrum 257 "S2" and "S3" for AISI 321, and "S6" and "S7" for AISI 347). The top-surface oxide layer on 258 metal surface under Ar atmosphere was less compact with evidence of a more severe spallation 259 than in air at 600 °C. The morphology and chemistry of corrosion oxides are also different (see 260 Table 2; spectrum "S3" and "S4" for AISI 321, and "S6" and "S8" for AISI 347). Similar trend 261 was also observed for In 825 and IN 625. Significant change in surface morphology of oxide 262 layers was observed on the IN 625 with change of gas atmosphere from air to Ar at 600°C, 263 shown in Figure 4 (c2) and (c3). For In 825 the corrosion oxide layer formed on sample under 264 Ar atmosphere at 600°C showed dispersed particles of corrosion oxides, while needle-like 265 corrosion oxides and crystalline oxide grains were randomly dispersed on the surface of In 825 266 after test in air. The top surface corrosion oxide layer was also observed to be more porous 267 when sample is exposed to Ar atmosphere than in air as shown in Figure 4 (d2) and (d3).

#### 268 3.4 Corrosion Oxide Characterisation; Cross -Sectional Surface Analysis

The cross-sectional analysis using the SEM was conducted to investigate the distribution of corrosion oxides across metal – oxide interface. This is presented in Figure 5.

271 Referring to Figure 5 (a1) (b1) (c1) and (d1), only a few corrosion oxides were identified from 272 EDS cross sectional analysis on samples' surfaces after immersion test in air at 565°C. This is 273 consistent with XRD patterns shown in Figure 3, and indicating low corrosion rate under this 274 condition. The thickness of oxide layers formed on stainless steels increased from ~ 3.5 to 10.5 275 µm with increase in temperature from 565 to 600°C in air. There is also evidence of detachment 276 and/or spallation of interfacial oxides in the form of an inner crack in Figure 5 (a2) (b2). When 277 samples are exposed to Ar atmospheres at 600°C, the corrosion oxide layer formed on AISI 278 321 is composed of two layers: an outer non-uniform layer of thickness of ~  $12 - 35 \mu m$ , and 279 an inner layer of thickness of  $\sim 13 \,\mu m$ . This is shown in Figure 5 (a3). The outer layer was 280 identified as Fe<sub>2</sub>O<sub>3</sub> and NaFeO<sub>2</sub> (referring to Figure 5 (a3) and Table 2, "S4"), while the inner 281 layer was a mixture oxide of (Fe, Cr):  $FeCr_2O_4$  spinel. Similar double oxide layers were 282 observed on AISI 347 with ~8  $\mu$ m thick outer Fe<sub>2</sub>O<sub>3</sub> layer and ~ 8  $\mu$ m thick inner FeCr<sub>2</sub>O<sub>4</sub> 283 layer, as shown in Figure 5 (b3). For IN 625, no significant differences were observed between 284 test at 565 and 600°C in air, while a 6  $\mu$ m thick double layer was observed on surface of sample exposed to Ar atmosphere at 600°C. This double oxide layer consists of an outer mixed oxide 285

of Fe and Ni, and inner oxide of Cr. For In 825, a thin inner oxide layer was observed after immersion test in air at 600°C, consisting of Fe<sub>2</sub>O<sub>3</sub> layer and some crystalline copper oxide (the copper from In 825 substrate itself in Table 1) dispersed on the top layer, as shown in Figure 5 (d2). The thickness of corrosion oxide layer formed on the In 825 samples under Ar atmosphere ranges from 9 -13  $\mu$ m, and also consist of an inner mixed oxides of Cr and Ni.

#### **291 3.5** Corrosion Oxide Characterisation – Raman Analysis

292 Raman measurement was used to identify the corrosion oxides and support the results from 293 XRD analysis. The Raman spectra and referred Raman shifts are shown in Figure 6 and Table 294 3. Figure 6 (a) and (b) indicates that the NaFeO<sub>2</sub> and Fe<sub>2</sub>O<sub>3</sub> forms as the outmost oxide layers 295 on stainless steels, while the inner layer consists of  $Cr_2O_3$  and  $FeCr_2O_4$  spinel. The corrosion 296 oxides formed on the surface of IN 625 are Cr<sub>2</sub>O<sub>3</sub> and a mixed oxide of Fe and Ni (identified 297 as NiFe<sub>2</sub>O<sub>4</sub> (NiO-Fe<sub>2</sub>O<sub>3</sub>)), shown in Figure 6 (c). The main corrosion oxides formed on the In 298 825 surface were identified as Fe<sub>2</sub>O<sub>3</sub>, Cr<sub>2</sub>O<sub>3</sub> and NiO-Fe<sub>2</sub>O<sub>3</sub> in Figure 6 (d). All of the Raman 299 spectra correlate well with the results from XRD analysis.

#### **300 3.6 Analysis of Stresses within Interfacial Corrosion Oxide Layers**

301 Table 4 lists the coefficients of thermal expansion (CTE) of some common oxides and the 302 studied alloys. However, due to the similarity in the chemistry of the corrosion oxide layers 303 and the variation in their thickness at different temperatures and test atmospheres, the samples 304 from test under Ar atmosphere at 600°C were used for the nano-indentation test. This was 305 because this condition provided samples with large enough thickness of the interfacial 306 corrosion oxide layers for this analysis. Figure 7 (a) shows the load-displacement (P-h) curves 307 obtained on the cross-sections of oxides formed on the alloys in this study. Based on the P-h 308 curves, hardness H and elastic modulus E of the base metal, inner corrosion oxide layer and 309 outer oxide layer were calculated using Oliver-Pharr method [32] and the results were plotted 310 in Figure 7 (b).

- The thermal stress values obtained using Equation 5 for the stress between inner layer and bulk
  material and between inner layer and outer layer were presented in Table 5. Referring to Table
  5 for calculated thermal stress values:
- 314  $\sigma_1$  represents the stress between the inner oxide layer and bulk substrate for samples under Ar 315 atmosphere at 600°C.

- 316  $\sigma_1$  represents the stress between inner oxide layer and outer oxide layer for samples under Ar
- atmosphere at 600°C.
- 318  $\sigma_2$  represents the thermal stress between inner oxide layer and bulk substrate for samples in air 319 at 565°C.
- 320  $\sigma_2$  represents the thermal stress between inner oxide layer and outer oxide layer for samples
- 321 in air at 565°C.
- 322  $\sigma_3$  represents the stress between the inner oxide layer and bulk substrate for samples in air at 323 600°C.
- 324  $\sigma_{3}$  represents the thermal stress between inner oxide layer and outer oxide layer for samples 325 in air at 600°C.
- 326 Details of calculation is shown in the supplementary files.
- 327
- 328



Figure 3 XRD patterns of studied alloys under different immersion conditions (phase number with underline meaning the 100% intensity reflections), (a) AISI 321, (b)AISI 347,
 (c) IN 625 and (d) In 825 after 7-day immersion test in Solar salt at (1) 565°C in air (2)600°C in air (3)600°C in argon, respectively (X-axis:2 theta in degree, y-axis intensity).



Figure 4: SEM top surface of the oxide scales formed on (a) AISI 321, (b)AISI 347, (c) IN 625 and (d) In 825 after 7-day immersion test in Solar
 salt at (1) 565°C in air (2)600°C in air (3)600°C in argon, note: the red scale bar on the inlays figure indicates 200 μm.

Element in wt%	Possible phase	Fe	0	Cr	Ni	Na
<b>S</b> 1	FeCr <sub>2</sub> O <sub>4</sub> , Fe <sub>2</sub> O <sub>3</sub>	60.1	18.9	12.1	-	0.1
S2	FeCr <sub>2</sub> O <sub>4</sub> , Fe <sub>2</sub> O <sub>3</sub>	50.1	22.9	20.6	4.4	0.5
<b>S</b> 3	NaFeO <sub>2</sub> , Fe <sub>2</sub> O <sub>3</sub>	54.7	20.2	1.4	0.2	22.6
<b>S</b> 4	NaFeO <sub>2</sub> , Fe <sub>2</sub> O <sub>3</sub>	49.99	32.4	0.62	0.32	6.73
S5	FeCr <sub>2</sub> O <sub>4</sub> , Fe <sub>2</sub> O <sub>3</sub>	69.8	17.9	12.1	10.1	0.2
<b>S</b> 6	NaFeO <sub>2</sub> , Fe <sub>2</sub> O <sub>3</sub>	60.7	21.8	1.2	0.6	13.8
<b>S</b> 7	FeCr <sub>2</sub> O <sub>4</sub> , Fe <sub>2</sub> O <sub>3</sub>	66.1	21.1	8.7	2.6	-
<b>S</b> 8	NaFeO <sub>2</sub> , Fe <sub>2</sub> O <sub>3</sub>	50.64	29.86	0.62	2.99	9.99
S9	FeCr <sub>2</sub> O <sub>4</sub> , Fe <sub>2</sub> O <sub>3</sub>	51.29	28.78	8.52	4.71	1.43
S10	NiO	2.6	7.1	16.2	54.0	-
S11	NiO	3.0	18.9	4.3	71.8	-
S12	NiO, (Fe, $Cr$ ) <sub>2</sub> O <sub>3</sub>	11.6	25.9	2.4	46.4	8.7
S13	NiO, (Fe, $Cr$ ) <sub>2</sub> O <sub>3</sub>	7.45	21.53	1.1	60.8	2.08
S14	NiO, (Fe, $Cr$ ) <sub>2</sub> O <sub>3</sub>	28.5	8.9	18.6	38.8	-
S15	NiO, Fe <sub>2</sub> O <sub>3</sub> ,NaFeO <sub>2</sub>	39.1	23.7	8.5	12.8	9.9
S16	NiO, Fe <sub>2</sub> O <sub>3</sub> ,NaFeO <sub>2</sub>	39.6	34.19	1.04	6.78	9.11

336 Table 2 EDS data for regions and possible phase in Figure 4



Figure 5: SEM-EDS profile of the oxide scales formed on (a) AISI 321, (b)AISI 347, (c) IN 625 and (d) In 825 after 7-day immersion test in Solar salt at (1) 565°C in air (2)600°C in air 340

(3)600°C in argon. 341



342

Figure 6 The detected Raman spectra for the development of the corrosion product scales
on studied alloys (a):AISI 321, (b):AISI 347, (c) IN 625 and (d) In 825 at various conditions.

346 Table 3 Raman frequencies (cm<sup>-1</sup>) of the oxides studied from literature [33-36]

Oxides	Raman shifts cm <sup>-1</sup>
Fe <sub>2</sub> O <sub>3</sub>	225, 290, 411, 501, 610
NaFeO <sub>2</sub>	220, 290, 350, 507
$Cr_2O_3$	353, 550, 616
FeCr <sub>2</sub> O <sub>4</sub>	686
NiFe <sub>2</sub> O <sub>4</sub>	487, 704



Figure 7: (a)Typical P-h curves of base material, inner layer and outer layer of AISI 321,
AISI 347, IN 625 and In 825 and (b) corresponding harness and elastic modulus (samples

350 AISI 347, IN 625 and In 825 and (
351 after 600°C isothermal in argon ).

Table 4 Linear coefficient of thermal expansion of metal and oxide [37] (temperature range
 100-800 °C)

Metal/oxide	Thermal expansion coefficient (*10 <sup>-6</sup> ) of Oxide [24]	Thermal expansion coefficient (*10 <sup>-6</sup> ) of metal [24]	Poisson ratio	
Fe-Fe <sub>2</sub> O <sub>3</sub>	12	15.2	0.25 [26]	
Fe -Fe <sub>3</sub> O <sub>4</sub>	15	13.5		
Ni-NiO	17.1	17.6		
Cr-Cr <sub>2</sub> O <sub>3</sub>	7.3	9.5	0.33 [26]	
Austenitic SS	-	16.0-19.5 (~17.4)	0.33	
Inconel 625	-	13.1-17.8 (~17.8)	0.308	

359 Table 5: Calculated thermal stress values (in GPa) between the various interfaces across the

360 oxide – metal interface for the different combination of alloys, temperatures and test

Motol	YS/TS <sup>#</sup>	Argon at 600°C		Air at 565°C		Air at 600°C	
Iviciai	At 600°C	σ1	σ'1	σ2	σ'2	σ3	σ'3
AISI 321	0.13/0.38	-1.335	0.158	-0.348	*	-1.335	0.169
AISI 347	0.16/0.36	-2.31	0.138	-0.229	*	-2.31	0.087
IN 625	0.34/0.80	-2.20	0.491	-0.118	*	-0.125	*
In 825	0.15/0.48	-1.045	0.013	-0.969	*	-1.032	*

361 *atmospheres investigated in this study.* 

362 \*: Not applicable with unavailable double layer.

363 <sup>#</sup>: YS and TS in the table stand for yield strength and tensile strength

## 364 **4. Discussion**

# 365 4.1 Effect of temperature on the corrosion characteristics and chemical evolution 366 of interfacial oxide layers.

Figure 2 shows the weight loss and corrosion rate of stainless steels and Ni-based alloys after
7-day immersion test in solar salt as a function of temperature and gas atmospheres (air and Ar)
respectively.

370 Referring to Figure 2 for all samples exposed to air, the corrosion rate increased with increasing 371 test temperature from 565 to 600°C. This trend shows good agreement with the literature [18]. 372 This is likely due to the synergy of effect of temperature in terms of enhanced aggression of 373 molten nitrate salts, and effect of temperature on changing the salt chemistry according to 374 Equation 1 and Equation 2. It has been reported that increased corrosiveness of solar salt is 375 likely from the decomposition of nitrate salt into nitrite salt and oxide ions when the salt's 376 stability limit of 565°C is surpassed [8, 38]. The equilibrium constant of Equation 1 and Equation 2 at 550°C is reported to increase with increasing temperatures [38], thus accelerating 377 378 the formation of oxide ions and expecting to enhance corrosivity of the salt.

From the XRD patterns and Raman spectra in Figure 3 and Figure 6 respectively, it is evident that the oxide layers formed on the surface of AISI 321 and AISI 347 in solar salt consists of different combination of iron oxide (Fe<sub>2</sub>O<sub>3</sub>), iron chromium spinel (FeCr<sub>2</sub>O<sub>4</sub>), sodium ferrite (NaFeO<sub>2</sub>), and chromium oxide (Cr<sub>2</sub>O<sub>3</sub>). For IN 625, nickel oxides (NiO) were identified on the corrosion surface after test in air, and spinel of Ni and Fe was also observed on the Raman 384 spectra. NiO and mixed oxides of (Fe, Cr) are both identified on the samples of In 825 after 385 test in air at 565 and 600°C. NaFeO<sub>2</sub> was identified on the In 825 after test at 600°C in air and 386 under Ar atmosphere (see Figure 3 (d2)). This is strong indication that the specific combination 387 of the interfacial oxide chemistries is directly dependent on the specific combination of test 388 temperature and gas atmosphere. This will also determine the distribution of thermo-389 mechanical stress across the corrosion interface, as well as the overall corrosion characteristics 390 of the molten salt – metal interface. The possible reactions pathway that guides the evolution 391 of these interfacial corrosion oxides have been reported in previous studies [11, 29, 39].

392 For AISI 321, 347 and In 825 tested in air, NaFeO<sub>2</sub> was identified with a higher peak intensity 393 at 600°C (see Figure 3 (a2), (b2) and (d2)). It is known to be porous, non-protective [39], and 394 with a high tendency to spall off as shown by region "S3" and "S6" in Figure 4 (a2) and (b2). 395 After spallation of NaFeO<sub>2</sub>, inner Cr-rich layer is then exposed to molten salt aggression as shown by region "S2" and "S7" in Figure 4 (a2) and (b2). This is a recipe for increased 396 397 corrosion activities at the interface, especially as it has been reported that Cr-containing oxide 398 was likely to be chemically dissolve into nitrate salt [40], therefore resulting in high corrosion 399 rate in Figure 2. The spallation behaviours were also observed on and confirmed from the 400 images from cross-sectional analysis in Figure 5 (a2) and (b2). It is also worth noting from 401 Figure 5 that a double oxide layer was observed on the surfaces of stainless steels after test at 402 600°C in air. This double oxide layer consists of an outer Fe<sub>2</sub>O<sub>3</sub>/NaFeO<sub>2</sub> and inner 403 FeCr<sub>2</sub>O<sub>4</sub>/Cr<sub>2</sub>O<sub>3</sub>. The initially formed dense and protective Cr<sub>2</sub>O<sub>3</sub>/ FeCr<sub>2</sub>O<sub>4</sub> layer prevents or 404 retards the outward diffusion of alloys elements from the metal and inwards diffusion of oxide 405 ions, while the outer Fe<sub>2</sub>O<sub>3</sub> can provide some level of obstacle to the dissolution of Cr-rich later 406 into molten salt. This synergy of barriers to continuous corrosion degradation mechanisms are 407 destroyed with the formation and spallation of NaFeO<sub>2</sub> at temperatures higher than 600°C [16], 408 therefore resulting in higher corrosion rate and deeper corrosion depth.

409 Referring to SEM and cross-sectional images shown in Figure 4 and Figure 5 (a1) - (d1), 410 relatively smooth surfaces were observed on the surfaces of all samples from test at 565°C in 411 air to indicate the formation of a thin, dense and compact corrosion oxide layer on top of 412 substrate metal. Polishing scratches are still visible on the surface of Ni-based alloy as evidence 413 of supressed corrosion activities. Mild corrosion occurred on the samples at 565°C in air. 414 However, with the increase in temperature, the corrosion process is greatly accelerated, acting in synergy with decomposed salt chemistries to drive new interfacial oxide formation andspallation.

# 417 4.2 Effect of test atmosphere on corrosion characteristics and chemical evolution 418 of interfacial oxide layers.

419 As briefly mentioned earlier, the corrosion rate of studied alloys increases significantly with 420 change in test atmosphere from air to argon at 600°C. There is a general belief that oxidation 421 and corrosion process should have been reduced in an inert protective atmosphere (argon for 422 example). The reduced access to oxygen at salt-metal interfaces will likely reduce corrosion 423 rate and localized attack due to inhomogeneity in the metal surface [16]. However, in this study, 424 the obtained corrosion rate and the visible corrosion surface indicates that the corrosion process 425 occurring under Ar atmosphere is more severe. This has been linked to a combination of 426 specific reaction pathways that influences the corrosiveness of solar salt, the impact of delayed 427 passivation from Cr<sub>2</sub>O<sub>3</sub> due to slow availability of oxygen, and preferential formation of other 428 non-protective oxide layers from previous publications [29, 41]. S. Bell [41] highlighted similar 429 complexity related to corrosion in oxyanionic salt environment, and believed that basicity, and 430 oxygen availability were responsible for the oxidation state of Fe and Cr, thus affecting the 431 corrosion process.

432 Referring to the XRD patterns in Figure 3 (a2), (a3), and (b2), (b3) for stainless steel samples, 433 Cr<sub>2</sub>O<sub>3</sub> can be identified on the sample surface at 600°C in air, but not in Ar atmosphere. It is 434 expected that Cr<sub>2</sub>O<sub>3</sub> forms as a passive oxide layer to protect the alloy element's outwards 435 diffusion. The absence and/or delay in the formation of  $Cr_2O_3$  on surface of samples under Ar 436 atmospheres could be explained by unavailability and/or slower rate of diffusion of oxygen 437 from air atmosphere. This has been observed in this study in the early stages of the evolution 438 of the corrosion interface, and corroborates the observations of S. Bell [41]. Delayed 439 passivation will also imply a non-equilibrium in favour of availability of Cr, Fe at the molten 440 salt – alloy interface. The outward diffusion of these atoms from bulk material to the corrosion 441 interface is driven by high temperature, resulting in high concentration of atoms at the interface. 442 The absence of a prompt supply of oxygen to promote quick passivation of the interface will 443 lead to the formation of thicker films of Cr-oxide layers as soon as oxygen becomes available 444 via decomposition of nitrate to nitrite ions. Meanwhile, due to the high affinity of Fe and O, 445 the formation and accumulation of Fe<sub>2</sub>O<sub>3</sub> would promote the formation of NaFeO<sub>2</sub> by the 446 Equation 6 (refer to supplementary file). This is shown by the higher peak intensity for NaFeO<sub>2</sub> in XRD patterns for test under Ar atmosphere than in air. Due to the porous and non-protective
nature of NaFeO<sub>2</sub> (shown in Figure 4), thicker corrosion oxide scales are likely to form with a
rather uneven thickness (as shown in Figure 5). The NaFeO<sub>2</sub> outer layer has a high tendency to
spall off from the interface.

451 While the formation and spallation of NaFeO<sub>2</sub> is also observed in air, the initially formed inner 452 Cr<sub>2</sub>O<sub>3</sub> limits the outward diffusion of Fe and formation of Fe<sub>2</sub>O<sub>3</sub>, and therefore decrease the 453 amount of NaFeO<sub>2</sub>. Despite the observation of spallation in Figure 4 (a2) (b2), the corrosion 454 oxide layer is thinner with a uniform thickness. The different source/pathways for the 455 availability of oxygen to the corrosion interface in air and under Ar atmospheres directly affects 456 the kinetics of  $Cr_2O_3$  formation and the evolution of other interfacial oxide chemistries. A 457 recent study confirmed [21] that the corrosion was accelerated when an inert argon 458 environment was used in chloride/sulphate salt for nickel superalloy based on the basic salt 459 corrosion mechanism and sodium metal oxides are favourable formed. In a previous study [29], 460 it can be concluded that the salt aggressiveness increases with the increase of nitrite and oxide 461 ions concentration by altering test atmosphere from air to Ar. Considering that the nitrite ions' 462 oxidizing potential is much lower than that of the nitrate ions, the enhanced salt corrisivity is 463 mainly caused by oxide ions from the complex decomposition of nitrate ions, as highlighted 464 by work from Sötz [42, 43].

# 465 4.3 Correlation between interfacial stress distribution and the corrosion 466 characteristics of interfacial oxides

467 In this study, the nano-mechanical properties of the interfacial corrosion oxide layers are 468 investigated using nano-indentation technique in order to explain the relationship between 469 spallation tendency of interfacial oxide layers and different stress levels across the metal -470 oxide interface. Referring to Figure 7, the hardness of base metal is higher than that of outer 471 oxide layer but lower than that of inner oxide layer for Stainless steels. This indicates that the 472 inner oxide layer is more dense, resilient and protective whilst outer oxide layer is porous, less 473 adherent and resilient. As shown in Figure 5 (a3) (b3), the EDS line and mapping show that 474 inner oxide layer is a Cr-rich layer and the porous outer Fe<sub>2</sub>O<sub>3</sub>/NaFeO<sub>2</sub> layer is characterised 475 with a high-density of defects (e.g. pores and micro-cracks). The high hardness and stiffness 476 associated with the continuous and compact inner oxide layer formed on the Stainless Steels 477 samples (under Ar atmosphere at 600°C) indicates that the inner oxide layer is capable of 478 providing an effective barrier to isolate the base metal from the oxidizing/corrosive molten salt 479 species. The high strength inner oxide layer also implies a higher resistance to cracking and/or breakdown. For Ni-based alloys, the hardness difference between inner/outer oxide layers with 480 481 base metal is not as much as that of Stainless Steels. The hardness of base metal is almost same 482 with that of outer oxide layer and slightly lower than that of inner oxide layer. Referring to the 483 cross-sectional images shown in Figure 5, Cr-rich inner oxide layer was observed on the 484 surfaces of Ni-based alloys, while outer Fe-rich oxide layer was also observed to have more 485 pores and micro-cracks. In comparison with the outer oxide layers formed on all the studied 486 alloys, the hardness of outer oxide layer on Ni-based alloys are almost 2 times higher than that 487 on the stainless-steel samples, indicating the outer oxide layer is more resilient on Ni-based 488 alloys.

489 As briefly discussed earlier in this paper, stress associated with the formation and growth of 490 oxides on Stainless Steels and Ni-based alloys maintains protective under compressive stress. 491 Referring to the thermal stress calculated from the Equation 5 (listed in Table 5), the negative 492 sign of thermal stress between inner oxide layer and base metal also shows the manifestation 493 of compressive stress for all the alloys; with values ranging from 1.045 - 2.31 GPa for samples 494 tested under Ar atmosphere at  $600^{\circ}$ C, 1.032 - 2.31 GPa for samples tested in air at  $600^{\circ}$ C, to 495 0.118 - 0.348 GPa for samples test in air at 565°C. This shows great agreement with the 496 observed higher strength of compact inner oxide layers and their relative resilience and 497 protectiveness. The tensile stress is imposed between inner oxide layer and outer oxide layers; 498 with values ranging from 0.013 - 0.491 GPa for samples tested under Ar atmosphere to 0.087 499 - 0.169 GPa for stainless steels samples tested in air at 600°C. The tensile stress between inner 500 and outer oxide layers discussed above were calculated for corrosion interface with double 501 oxide layers; This was not applicable for the samples in air at 565°C or the Ni-based alloys 502 samples at 600°C in air (see Table 5).

503 These results clearly show that the compressive stress (negative sign) helps to keep the inner 504 oxide layer adhered to base metal and tensile stress (positive sign) leads to spallation of outer 505 oxide layer. The reasons responsible for the results above are probably attributed to the 506 structure of corrosion oxide layers, and the fact that inner oxide layer is initially formed as the 507 primary corrosion oxides with higher hardness and strength, and better adhered to the base 508 material. For stainless steel, the outer oxide layers (identified as NaFeO<sub>2</sub> by SEM) show lower 509 hardness than inner oxide layer. This is also confirmed from the stress calculation to be of poor 510 adherence to the inner Cr-rich oxide layer, and hence a recipe potential spallation to occur. The 511 tensile stress or thermally induced stress obtained between inner oxide and outer oxide layers 512 could also be partly attributed to the thermal expansion coefficient of inner Cr-based oxide 513 layer and outer Fe-based oxide layer. Therefore, the dense, more adherent and resilient inner 514 oxide layer is less likely to spall off by thermo-mechanical stress effects after 7 days. However, 515 this does not completely prevent the chemical attack of the inner oxide layer by aggressive 516 molten salt species, particularly after the spallation of the outer oxide layers. As shown in 517 Figure 5, the inner oxide layer was associated with less spallation after 7 days of immersion 518 test in solar salt. It is believed from this study that the tendency for spallation of interfacial 519 oxide layers to occur on the surface of these alloys is closely related to mechanical resilience 520 of the interfacial oxide layers.

## 521 **5.Conclusions**

522 This study has successfully investigated the early stages (over 7 days) of corrosion behaviour 523 of various alloys as a function of temperature and test atmosphere with the aim of correlating 524 the overall corrosion characteristics at the molten salt – oxide – metal interface with the stress 525 distribution across the interface. Here are the key conclusions.

526

527 1. High temperature and Ar test atmosphere acts in synergy to increase the corrosion rate up 528 to ~3 fold for all alloys tested at 600°C This synergy is related to a combination of molten 529 salt induced electrochemical activities, Arrhenius effect of high temperature on the rate of 530 decomposition of nitrate salts to nitrite and delayed passivation by  $Cr_2O_3$  due to lower 531 availability of oxygen at the corrosion interface under Ar atmosphere.

532

533 2. The gas atmospheres and temperature synergistically controlled the overall corrosion
534 characteristics at the molten salt – oxide – metal interface by its effect on the evolution of
535 multiple interfacial corrosion oxide layers. Porous and less protective outer oxide layer
536 consisting of a mixture of NaFeO<sub>2</sub> and Fe<sub>2</sub>O<sub>3</sub> is kinetically favoured to form preferentially
537 at 600°C under argon atmosphere.

538

539 3. The rate and sequence of evolution of interfacial oxide layers of varying chemistries
540 directly correlate to the thermo-mechanical stress profile across the molten salt – oxide –
541 metal interface. This also directly correlates with the tendency for spallation to occur and
542 overall corrosion behaviour.

- 544 4. The spallation of sodium ferrite (NaFeO<sub>2</sub>) formed on samples tested in air and under argon
  545 atmosphere at 600°C directly correlates to the tensile stress between the Cr-rich inner oxide
  546 layer and the NaFeO<sub>2</sub> layer.
- 547
- 548 5. The observed inner Cr-rich oxide layer is more adherent and resilient to degradation than 549 the outer Na/Fe-based oxides. This was observed in this study to be due to the compressive 550 stress between the inner Cr-rich oxide layer and the substrate material, and the destructive 551 tensile stress between inner Cr-rich outer layer and outer Na/Fe-rich oxides layer to drive 552 spallation process.
- 553 Dedicated to Late Prof. Anne Neville. OBE, FREng, FRS, FRSE, RAEng Chair in
- 554 Emerging Technologies, University of Leeds.

# 555 Declaration of competing interest

556 The authors declared that they have no conflicts of interest to this conflict of interest to this 557 manuscript.

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# 561 Data Availability

562 Data will be made available on request.

# 563 **CRediT authorship contribution statement**

564 Qingyang Liu: Conceptualization, Methodology, Formal analysis, Investigation, Resources, Writing - original draft. Chun Wang: Supervision, Data collation, Methodology, Writing -565 566 & editing. Jiong **Qian**: Review & editing. Neville: Resources. review Anne 567 Supervision. Frederick Pessu: Supervision, Writing - review & editing, Resources, 568 Conceptualization.

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