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# Correction of higher mode Pochhammer–Chree dispersion in experimental blast loading measurements

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## Abstract

Experimental measurements of blast loading using Hopkinson pressure bars are affected by dispersion which can result in the loss or distortion of important high-frequency features. Blast waves typically excite multiple modes of propagation in the bar, and full correction of dispersive effects is not currently possible as the magnitude of stress propagating in each mode is not known. In this paper we develop an algorithm for multiple mode dispersion correction based on rigorous interrogation of the results from a series of finite element analyses. First, a finite element model is validated against first-mode and higher-mode theory. The dispersion of short raised-cosine windowed pulses is then used to isolate the contribution of each propagating mode, enabling a relationship between frequency and modal apportioning of stress to be obtained for the first four propagating modes. Finally, four-mode dispersion correction is successfully applied to an experimental signal using an algorithm based on the derived relationships for modal apportioning. The four-mode results show significant improvement in the capture of high-frequency features over existing first-mode corrections, and demonstrate the potential of this method for the full correction of dispersion in experimental measurements of blast loading.

*Keywords:* Dispersion correction, Hopkinson pressure bar, Multiple mode, Pochhammer-Chree, LS-DYNA

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## 1. Introduction

The provision of adequate blast protection systems is dependent on our ability to accurately quantify the loading arising from detonation of a high explosive. Blast events present a considerable challenge to the experimentalist: measurement devices should be robust enough to withstand pressures in the range of 10s–100s MPa, whilst being sufficiently sensitive and fast-acting to resolve temporal features in the kHz–MHz range.

The Hopkinson pressure bar [1], or HPB, is commonly used as a dynamic force transducer in investigations of the loading from explosive events, as it combines the durability and sensitivity required to record the high-pressure

26 transient features inherent in blast and shock waves [2–11]. In a HPB, a stress wave applied to one end of the  
27 bar propagates down its length, and is recorded by strain gauges fixed to the bar surface some distance away. The  
28 wave equations derived by Pochhammer [12] and Chree [13] indicate that each frequency component of a stress  
29 wave will propagate at a velocity dependent on the frequency of that component. Thus, a signal will disperse as it  
30 propagates from the loaded face of the pressure bar to the strain gauge.

31 Dispersion of a stress wave is of particular concern in the measurement of explosive events, as it can result in  
32 the distortion or loss of important high-frequency features, impeding accurate quantification of the loading [14].  
33 Dispersion of the stress wave is also accompanied by a frequency-dependent variation in stress across the bar  
34 cross-section, and so a signal recorded on the bar surface may not be representative of the mean stress in the bar  
35 [15–17].

36 The Pochhammer–Chree equations have an infinite number of roots corresponding to different modes of prop-  
37 agation, so that a high-frequency component may propagate in more than one mode, each with its own phase and  
38 group velocities (Figure 1). The transmission of energy in the bar occurs at the group velocity,  $c_g$ , while individual  
39 wave features, such as a particular crest, propagate at the phase velocity,  $c_p$ . While the phase velocity in the higher  
40 modes often exceeds  $c_0$  (Figure 1a), the energy associated with these waves will always propagate in the bar at  
41 velocities below  $c_0$  (Figure 1b).

42 When only the first mode of propagation is excited, the frequency domain dispersion correction method de-  
43 veloped by Gorham [18] and Follansbee and Frantz [19] can be used. In this method, the fast Fourier transform  
44 (FFT) of a signal is taken, the phase angle of each frequency component is adjusted according to the phase velocity  
45 derived from the Pochhammer–Chree equations, and an inverse FFT returns the signal to the time domain. Tyas  
46 and Watson [20] extended this method to account for the variation of stress across the bar cross-section by ap-  
47 plying additional corrections to the amplitude of each frequency component, an approach which was later verified  
48 experimentally by Tyas and Pope [21].

49 Above the cutoff frequency for the second mode of propagation, at least two modes will propagate at any  
50 given frequency, and each mode will propagate with a distinct velocity and amplitude. In signals which contain  
51 significant high frequency content, such as blast and shock waves, the magnitude of each frequency component  
52 therefore represents the sum of the stresses propagating in each mode at that frequency. If the portion of the  
53 total stress assigned to each mode is known, the amplitude of each frequency component can be divided amongst

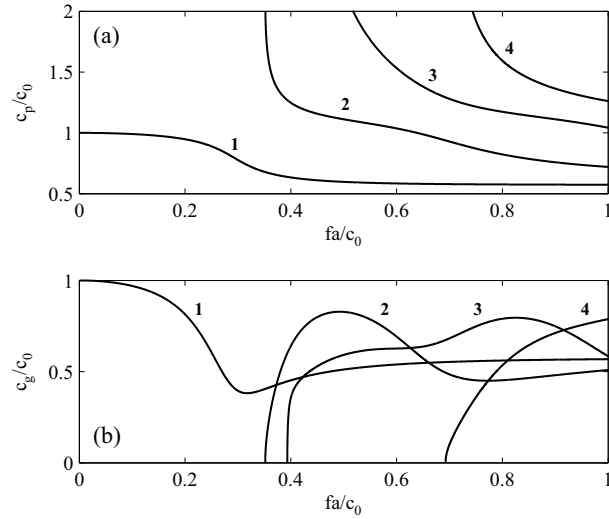


Figure 1: Variation of a) phase velocity,  $c_p$ , and b) group velocity,  $c_g$ , with frequency for the first four modes of wave propagation in a cylindrical bar ( $\nu = 0.3$ ).

54 the propagating modes, and then each mode individually corrected for dispersion according to its phase velocity.  
 55 However, this modal apportioning relationship is not provided by the Pochhammer–Chree curves, and so has been  
 56 a topic of research for many years.

57 Gregory and Gladwell [22] analysed the Pochhammer–Chree equations with an integral formulation of least  
 58 squares to determine the proportion of a sinusoidal wave propagating in each mode at normalised frequencies  
 59 up to  $fa/c_0 \approx 0.40$ , where  $f$  is frequency,  $a$  is the radius of the bar and  $c_0$  is the one-dimensional wave speed.  
 60 Puckett [23] used this approach to develop an analytical model for wave propagation in cylindrical bars, although  
 61 comparisons with experimental signals showed that the analytical model could not accurately predict dispersion  
 62 for wavelengths exceeding the bar diameter, i.e. when higher dispersion modes were excited.

63 Lee et al. [24, 25] used experimental pressure data with a Gaussian-windowed Fourier transform to estimate  
 64 the time of arrival and relative power of each frequency component in a blast wave. The results suggested that  
 65 at a given frequency the mode with the highest group velocity carried the majority of the signal energy. This led  
 66 to the development of dispersion curves based on the simplification that the entire magnitude of each frequency  
 67 component only propagates in the mode with the highest group velocity. However, Husemeyer [26] performed  
 68 multiple-mode dispersion simulations which showed that the contribution of the remaining propagating modes is  
 69 not sufficiently small to enable them to be neglected in this manner.

70 Despite a number of previous studies investigating multi-mode dispersion correction, there is no consensus  
71 and no proven method to estimate the magnitude of a given frequency component propagating in a given mode,  
72 other than for frequencies where only one mode propagates. No researchers to date have successfully isolated  
73 each mode of propagation, and the contribution of each mode to a reconstructed (dispersion corrected) pressure  
74 signal has yet to be demonstrated. The safe, economical design of blast-resistant structures, the investigation of  
75 explosive phenomena at small scaled distances, and the rigorous validation of numerical modelling approaches all  
76 depend on the development of reliable methods for the correction of higher-mode dispersive effects in experimental  
77 measurements of blast loading.

78 This paper presents a new method for correcting for dispersion in the first four modes of propagation. First,  
79 a finite element model is validated against theoretical predictions of the cross-sectional distribution of stress for  
80 single and multiple-mode propagation in a cylindrical bar. The finite element model is then used to derive ‘modal  
81 apportioning factors’ by propagating a windowed single-frequency signal down a HPB and recording amplitudes  
82 some distance down the bar, after the signal has separated into distinct modes. Finally, we present modifications to  
83 the Tyas and Watson [20] phase angle correction method to account for higher modes, and we present an example  
84 of higher-mode dispersion correction of an experimental signal. The contribution of each of the first four modes is  
85 calculated up to  $fa/c_0 = 0.90$ , enabling the correction of multiple-mode dispersion in experimental recordings of  
86 the loading from explosive events.

## 87 **2. Finite element model validation**

### 88 *2.1. Initial considerations*

89 The Pochhammer–Chree equations arise from the application of traction-free boundary conditions to the wave  
90 equation,

$$\frac{\partial^2 u}{\partial t^2} = c^2 \nabla^2 u. \quad (1)$$

91 In a finite element (FE) model of a cylindrical pressure bar, wave propagation is described by this same equation,  
92 and so complex phenomena such as multiple-mode propagation, dispersion and cross-sectional variations in stress  
93 should arise as predicted by the analytical methods. Finite element methods can therefore be particularly useful  
94 for investigating phenomena where the governing relationships are intractable, as in the case of modal stress  
95 apportioning.

96 Results can be obtained over the whole volume and time of interest, as the discretisation allows, however there  
97 are known limitations of the FE method which should be considered. The Pochhammer–Chree equations were  
98 derived for the case of an infinitely-long pressure bar subjected to an infinitely-long sinusoidal axial disturbance.  
99 While it is only possible to model an *approximation* of an infinitely-long pressure bar, the errors introduced by  
100 deviating from a *true* infinite bar can be minimised by windowing the forcing functions to reduce the inertial  
101 effects<sup>1</sup> introduced by the sudden application of stress at a free face [20, 26], or by enforcing zero displacement  
102 boundary conditions for a small length of bar equal to the wavelength of the forcing function [28].

103 Representation of mass in either a lumped or consistent matrix will give rise to spurious oscillations [29],  
104 and the central-differencing explicit time-stepping algorithm used in most finite element solvers will give rise to  
105 oscillations that are indistinguishable from Pochhammer–Chree dispersion [28]. The results from a finite element  
106 model may therefore begin to diverge from theory, particularly when considering cumulative numerical losses  
107 associated with propagating higher frequencies (relative to mesh size) over long distances. However, these errors  
108 can be minimized by using smaller, uniformly-sized elements and a time-step that is close to the critical time-step,  
109 given as the smallest element length divided by the one-dimensional wave speed.

110 To ensure that the modal apportioning of stress is accurately calculated by finite element methods, this section  
111 first compares the dispersion and radial distribution of stress modelled in LS-DYNA with the theoretical relation-  
112 ships, for both single and multiple mode propagation. The distribution of stress between the propagating modes  
113 is a result of the same wave equation responsible for these effects, and so accuracy in these behaviours is a good  
114 indicator that modal apportioning is also well captured.

## 115 2.2. Propagation in the first mode

116 Simulations were first compared to the Pochhammer–Chree relationships for situations where the forcing fre-  
117 quency was below the Mode 2 cutoff frequency ( $fa/c_0 \approx 0.35$ ), where it is known that only the first mode will  
118 propagate. A cylindrical steel pressure bar was modelled in LS-DYNA [30] with volume-weighted axi-symmetric  
119 shell elements and the *\*MAT\_ELASTIC* material model, using the parameters in Table 1 which were informed by  
120 previous mesh sensitivity studies [31, 32]. The dispersion of continuous sinusoidal waves of unit amplitude was

---

<sup>1</sup>When a pulse is applied suddenly to the end of a HPB the majority of the energy propagates axially along the bar. However, a small fraction of the energy remains at the loaded end of the bar in a very narrow frequency band. The energy of this resonant mode gradually leaks into the HPB as a normal propagating mode, known as an ‘evanescent mode’ [27].

	First mode	Higher modes
Element length, mm	0.2	0.4
Bar radius, mm	5	10
Bar length, m	3	10

(a)

Parameter	Value
Young's modulus, GPa	210
Density, Mg/m <sup>3</sup>	7.850
Poisson's ratio	0.3

(b)

Table 1: a) Mesh geometry and b) LS-DYNA *Elastic* material model parameters used for analysis of first and higher mode propagation.

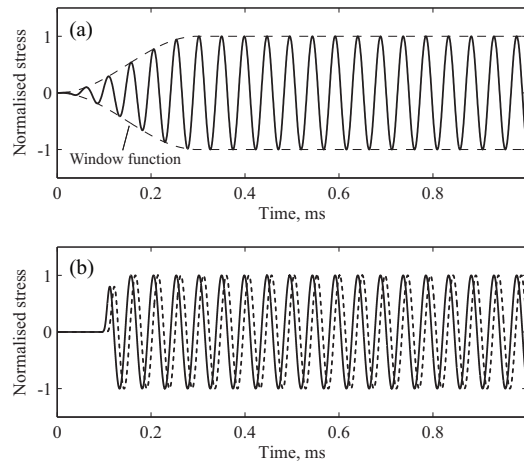


Figure 2: a) Sinusoidal forcing function with half raised-cosine window and b) stress measured on the bar axis 500 mm and 550 mm from the free end, for  $f a / c_0 = 0.02$ .

121 then assessed by modelling forcing functions at normalised frequencies up to 0.34 and comparing the resultant  
 122 phase velocities and cross-sectional variations of stress with those predicted by Pochhammer–Chree theory.

123 In each case, a sinusoid of unit amplitude was applied to the end of the bar for the entire 1 ms duration of  
 124 the analysis, representing an infinite duration pulse, with the addition of a 0.3 ms raised-cosine window applied  
 125 at the beginning of the wave to reduce the inertial effects as discussed above. To ensure that any further inertial  
 126 effects associated with the head of the applied pulse did not influence the results, results from the early stages of  
 127 the analysis were omitted. Only the latter stages of the output signals were considered, defined as the region over  
 128 which the stress response has constant peak amplitude.

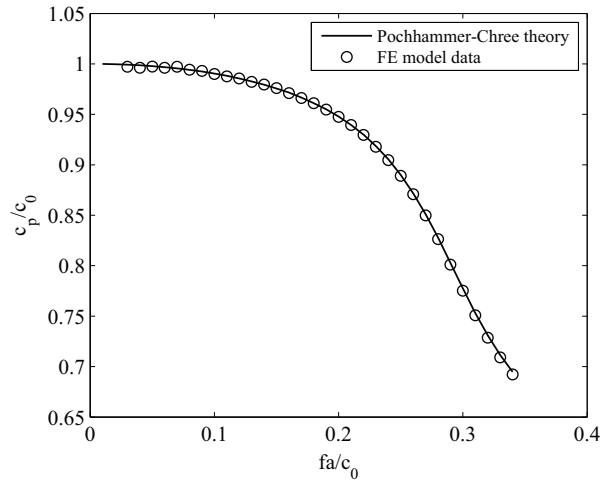


Figure 3: Relationship of phase velocity and frequency for Mode 1: comparison of Pochhammer–Chree theory and FE-modelled pressure bar.

129 Examples of the forcing function and the stress measured on the bar axis 500 mm and 550 mm from the end  
 130 are shown in Figure 2. Stresses were calculated by converting the axial component of nodal velocity ( $v$ ) into axial  
 131 stress ( $\sigma$ ) using the expression  $\Delta\sigma = \rho c_0 \Delta v$ . Phase velocity was calculated from these measurements by assessing  
 132 the change in phase angle that occurred between the two measurement locations. Phase velocities for normalised  
 133 frequencies up to 0.34 are shown in Figure 3, where it can be seen that the numerical results closely match the  
 134 relationship derived by Davies [15] through manipulation of the Pochhammer–Chree equations.

135 The distribution of stress across the bar cross-section was investigated in a similar manner for a number of  
 136 normalised frequencies between 0.09 and 0.34 by recording the stress at various radial ordinates between the bar  
 137 axis and surface at a location 500 mm from the end of the bar. The measured stresses were normalised against the  
 138 stress at the bar axis. The results are shown in Figure 4, where again it can be seen that the model is in very good  
 139 agreement with the Davies [15] theoretical values at each normalised frequency.

### 140 2.3. Propagation in higher modes

141 In order to investigate wave propagation in higher modes, the radius and length of the pressure bar were  
 142 increased (Table 1). This allowed lower velocity frequency components to reach the monitoring location without  
 143 any reflected waves arriving from the distal end of the bar and corrupting the results. Thus, the assumption of an  
 144 infinite-length bar was maintained. A full raised-cosine window was applied to a sinusoid of the required frequency  
 145 to produce a pulse of 0.2 ms length, as shown in Figure 5a. The wave envelope created by the windowing function



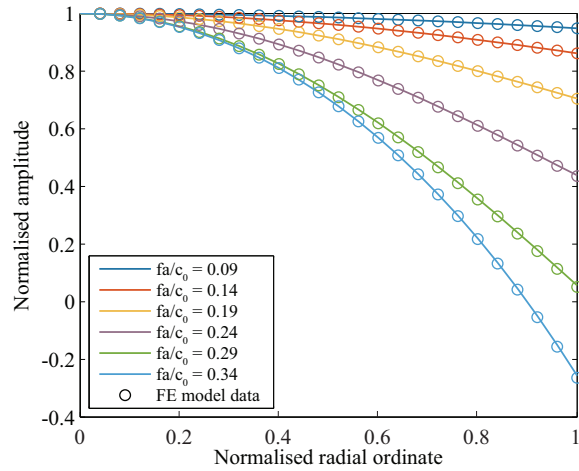


Figure 4: Spatial distribution of stress over the bar cross-section for Mode 1: comparison of Davies' analysis [15] and FE-modelled pressure bar.

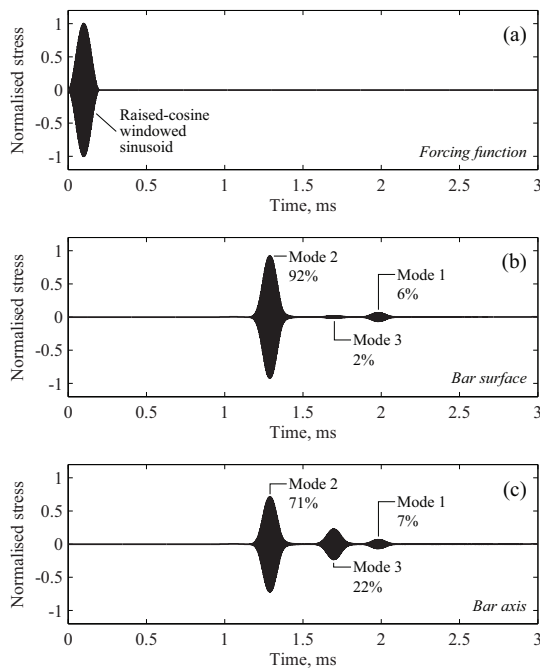


Figure 5: a) Raised-cosine windowed sinusoidal forcing function, and stresses measured after 6.0 m propagation on the bar b) surface and c) axis, for  $fa/c_0 = 0.52$ .

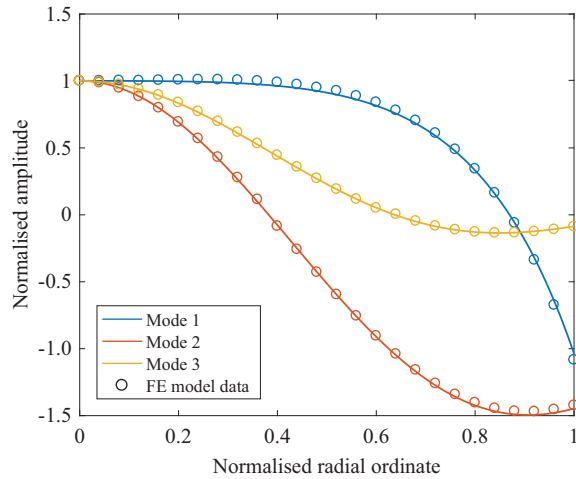


Figure 6: Spatial distribution of stress over the bar cross-section for Modes 1–3 at  $fa/c_0 = 0.52$ : comparison of Davies’ analysis [15] and FE-modelled pressure bar.

146 propagates in the bar at the group velocity, and so, as each mode propagates at its own group velocity, a single-  
 147 frequency input pulse will disperse and eventually separate into several distinct pulses, equal to the number of  
 148 modes present. This effect is shown in Figures 5b and 5c for a sinusoid with a normalised frequency of 0.52, where  
 149 the signal propagates in Modes 1–3. As a finite-duration input signal was used, at a point 6 m down the bar three  
 150 separate modes can be clearly distinguished and identified using measurements of group velocity. From this, the  
 151 relative amplitude of the pulses can be calculated.

152 The stress propagating in each mode at a particular radial ordinate was taken as the peak value of each distinct  
 153 modal pulse, and by recording at a variety of radial ordinates, the distribution of stress over the bar cross-section  
 154 was calculated for each mode. The cross-sectional distributions arising from the forcing function in Figure 5a are  
 155 shown in Figure 6; these results remain close to the theoretical values obtained from Davies’ analysis [15] even  
 156 though multiple modes are now propagating. This suggests that while numerical errors may be present (Section  
 157 2.1), they are insufficient to cause significant deviation from analytical theory. Hence, as LS-DYNA can replicate  
 158 these dispersive and radial stress distribution effects in waves propagating in higher modes, we can be confident  
 159 that the proportion of stress associated with each individual mode is also being modelled correctly.

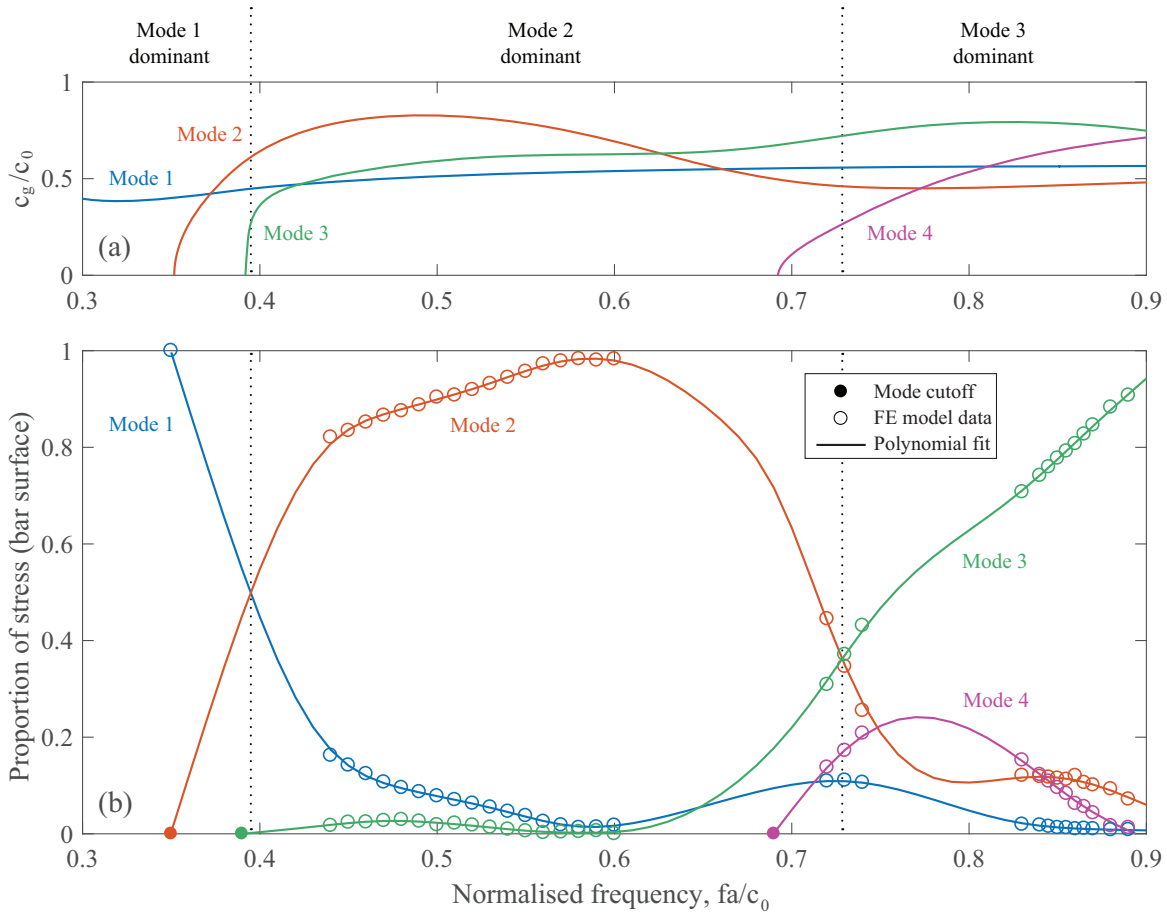


Figure 7: a) Group velocities of Modes 1–4 and b) the proportion of stress travelling in each mode on the bar surface, for  $0.30 < fa/c_0 < 0.90$ .

### 160 3. Derivation of modal apportioning factors

#### 161 3.1. Modal apportioning factors and graphical representation

162 Using the method in Section 2.3, the modal distribution of stress measured on the surface of the bar was  
 163 assessed at a range of normalised frequencies up to 0.90, at which point four modes are propagating simultaneously.  
 164 The results are shown in Figure 7, where the abscissa denotes the modelled frequency for each separate numerical  
 165 analysis, and the ordinates represent the contribution of each mode at that frequency. As an example, at  $fa/c_0 =$   
 166 0.52 the proportion of stress propagating in Modes 1, 2 and 3 is 6%, 92% and 2% respectively, as was previously  
 167 shown in Figure 5.

168 At frequencies where the group velocity of two or more modes is similar, the relative dispersion is not large  
169 enough for the clear separation of the forcing function into modal pulses. As it is not possible to identify the con-  
170 tribution of individual modes in these cases, the modal stress distribution cannot be resolved at these frequencies  
171 using the current method. However, an indication of the behaviour in these regions can be provided using the  
172 piecewise polynomial fits shown in Figure 7. Two additional constraints are applied to the fits: the stress propa-  
173 gating in a particular mode is zero below the cutoff frequency of that mode; and the proportions of the propagating  
174 modes should sum to unity at any normalised frequency. Any error initially present in the sum of the modes was  
175 minimised through iterative distribution between the contributing modes and refitting of the curves, so that even  
176 outside the modelled results the error in the sum of the fits remains below 1%. The modal proportions in Figure 7  
177 are provided in a supplementary file [33] as the ratios  $r_1$ ,  $r_2$ ,  $r_3$  and  $r_4$ , along with the cross-sectional mean for each  
178 mode,  $\bar{r}_1$ ,  $\bar{r}_2$ ,  $\bar{r}_3$  and  $\bar{r}_4$ , to enable future investigation and development of further correction algorithms. Results  
179 are provided at normalised frequency increments of 0.01, and indicate where the modal contributions are derived  
180 directly from the modelling study ('M'), or interpolated from the piecewise polynomial fits ('I').

### 181 3.2. Interpretation

182 The modal apportioning of stress on the bar surface in Figure 7 indicates a gradual transfer of stress from  
183 Mode 1 to Mode 2 at frequencies above the Mode 2 cutoff. From approximately  $fa/c_0 = 0.40$  the majority of the  
184 stress propagates in Mode 2, and this remains the case until  $fa/c_0 = 0.73$ , where Mode 3 becomes dominant. Mode  
185 4 does not become dominant over the considered frequency range. These results appear to be in good qualitative  
186 agreement with the energy partitions of Lamb waves derived in the case of plane strain response of semi-infinite  
187 elastic strips to harmonic excitation [34].

188 Comparison of Figures 7a and 7b indicates that the frequencies over which a particular mode is dominant do  
189 not closely relate to the frequencies at which that mode has the highest group velocity, as was reported by Lee  
190 et al. [25]. For example, the group velocity of Mode 3 exceeds all other propagating modes from  $fa/c_0 = 0.63$ ,  
191 but the proportion of stress propagating in Mode 3 does not exceed that of the other modes until  $fa/c_0 = 0.73$ .  
192 Between these frequencies Mode 2 is also briefly the *slowest* propagating mode while remaining dominant in  
193 terms of stress proportion. These results additionally confirm that the magnitudes of the non-dominant modes are  
194 not small enough to justify neglecting their contributions, except within narrow frequency bands, as discussed by  
195 Husemeyer [26].

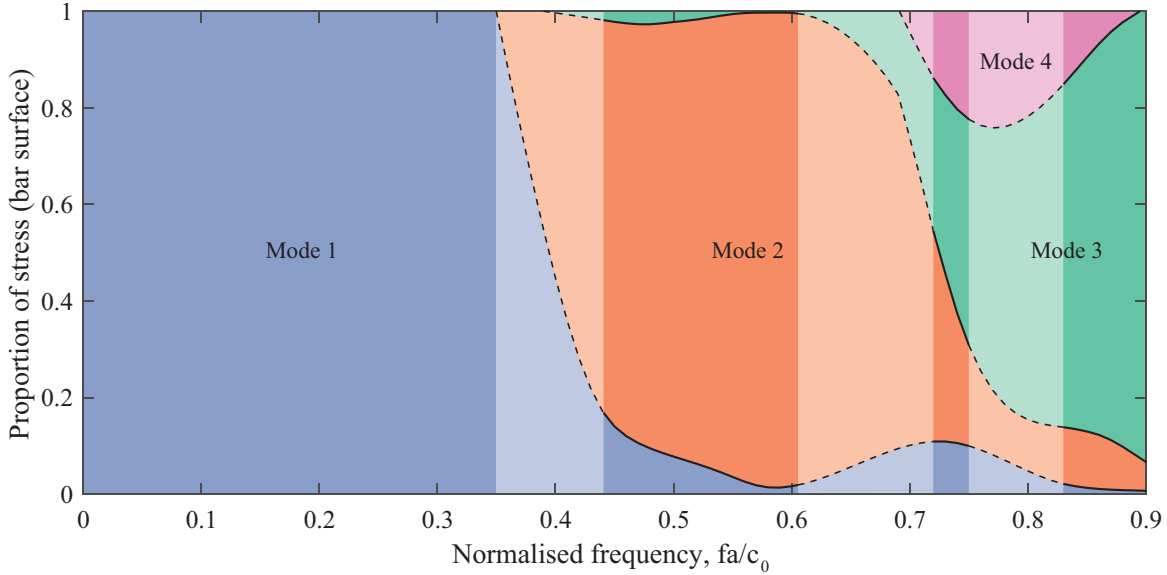


Figure 8: Distribution of stresses across Modes 1–4 for  $0 < fa/c_0 < 0.90$ . Dashed lines indicate interpolated data.

196 The data in Figure 7 can also be represented using stacked areas as in Figure 8, which shows how the total stress  
 197 associated with each propagating frequency component is distributed between the first four modes for normalised  
 198 frequencies between 0 and 0.90. In this figure the greyed-out areas represent regions where the group velocities  
 199 of the propagating modes are similar and the modal apportioning relationships have been interpolated from the  
 200 piecewise polynomial fits, as discussed previously. Successful measurements at these frequencies would require a  
 201 much longer pressure bar to enable sufficient relative dispersion. This has not been attempted in the current work  
 202 as the additional length would have introduced significant alteration of the signal through numerical losses.

203 The modal distribution of stresses in Figure 8 can be used with phase angle adjustments to calculate the stress  
 204 acting at another point on the bar surface, as in the following example. The next challenge in higher mode dis-  
 205 persion correction will be to consider the radial variation of stress and strain in the bar, using factors  $M_1$  and  $M_2$   
 206 for each propagating mode [20]. The values of  $M_1$  and  $M_2$  can be easily derived for any mode, but contain dis-  
 207 continuities at frequencies where nodal cylinders coincide with the bar surface, and so further investigation will be  
 208 required to quantify the data lost due to these nodal cylinders and develop best practice on their use.

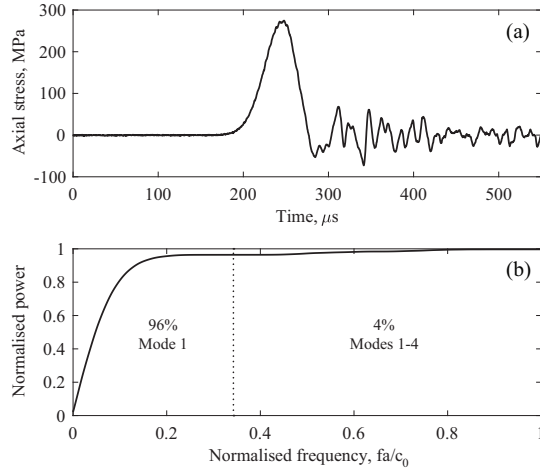


Figure 9: a) A dispersed experimental Hopkinson pressure bar signal and b) the normalised cumulative power with frequency.

#### 209 4. Higher mode correction of an experimental signal

210 The numerical results above indicate that it is now possible to determine how a signal will separate into its  
 211 component modes as it propagates in a Hopkinson pressure bar: this can be demonstrated by applying higher  
 212 mode correction to an experimental signal. Figure 9a shows a signal recorded 1 m from the end of an explosively-  
 213 loaded Hopkinson pressure bar used to perform acceleration tests on electronic components at the University of  
 214 Sheffield Blast & Impact Laboratory in Buxton, UK. The steel pressure bar is 2 m long, 100 mm diameter and  
 215 has  $c_0 = 5185$  m/s and  $\nu = 0.3$ . The dispersion of the signal is clear from the tail of high-frequency oscillations  
 216 following the main loading: these oscillations would continue beyond the  $550 \mu\text{s}$  shown here, but are overwritten  
 217 by the stress wave reflecting from the end of the bar. The cumulative power plot in Figure 9b, determined from an  
 218 FFT of the recorded signal, indicates that 96% of the signal is at frequencies below the Mode 2 cutoff, and so will  
 219 propagate in Mode 1 only. The remaining 4% is above the Mode 2 cutoff, and will propagate in two, three or four  
 220 modes simultaneously.

221 Correction of this signal was performed to obtain the stress on the surface of the bar at the loaded face, removing  
 222 the dispersion associated with 1 m of travel. A first-mode dispersion correction program [35, 36] was modified to  
 223 incorporate the proportion of stress propagating in each of the first four modes, using the values for the bar surface  
 224 derived in Section 3. The original program uses the method described by Tyas and Watson [20] to apply a phase

225 angle correction to each frequency component,  $x(\omega)$ ,

$$x(\omega) = A(\omega) e^{i\phi(\omega)} \quad (2)$$

226 where  $A(\omega)$  is the amplitude of the component and  $\phi(\omega)$  is the phase angle, and both are functions of the compo-  
 227 nent's angular frequency  $\omega$ . The phase angle correction,  $\phi'(\omega)$ , is calculated as

$$\phi'(\omega) = \left( \frac{c_0}{c_p(\omega)} - 1 \right) \frac{\omega z}{c_0} \quad (3)$$

228 where  $z$  is the propagation distance to correct over. A modified frequency component,  $x'(\omega)$ , can then be recon-  
 229 structed as

$$x'(\omega) = A(\omega) e^{i(\phi(\omega) - \phi'(\omega))}. \quad (4)$$

230 To incorporate Modes 2, 3 and 4, phase angle corrections ( $\phi'_1(\omega)$ ,  $\phi'_2(\omega)$  etc.) are calculated for each mode ac-  
 231 cording to the phase velocity at that frequency. The modified frequency component is then given as the sum of the  
 232 portions propagating in each mode, which are scaled by the mode ratios ( $r_1(\omega)$ ,  $r_2(\omega)$  etc.) derived in Section 3:

$$x'(\omega) = \sum_{m=1}^4 r_m(\omega) A(\omega) e^{i(\phi(\omega) - \phi'_m(\omega))}. \quad (5)$$

233 Figure 10 shows a comparison of one-mode and four-mode dispersion correction of the signal in Figure 9a,  
 234 where corrections have been applied to remove the dispersion associated with 1 m of travel in the bar. 'One-mode'  
 235 correction assumes that the total signal at each frequency propagates only at the Mode 1 phase velocity: this is the  
 236 current limitation imposed on dispersion correction without the modal apportioning factors derived in this paper.  
 237 While only 4% of the power of the signal is propagating in multiple modes, there is a significant difference in the  
 238 loading predicted by the two methods.

239 One-mode correction of the signal reduces the tail of high-frequency components following the main stress  
 240 wave, but, as all frequencies are assumed to propagate at Mode 1 phase velocities, large oscillations are intro-  
 241 duced ahead of and during the signal. For the current signal, one-mode correction remains adequate for non-local  
 242 measurements, such as total impulse, as the large local errors largely cancel each other out over over the range of  
 243 interest. For example, the uncorrected signal in Figure 10b has a total impulse of 95.2 N s, while the one-mode and

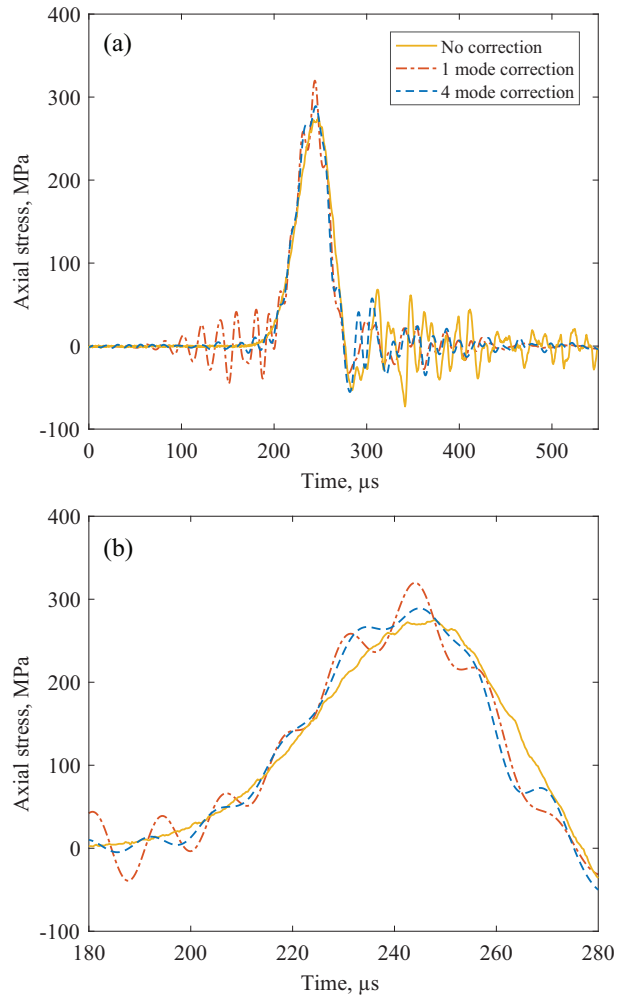


Figure 10: Single-mode and multiple-mode dispersion correction of a Hopkinson pressure bar signal, showing a) the complete signal and b) the main loading wave in detail. The 'no correction' signal is represented by the original recorded signal timeshifted using the one-dimensional wave speed of the bar,  $c_0$ .



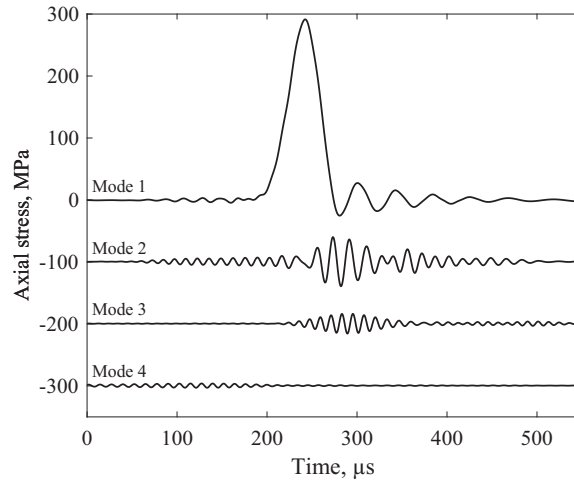


Figure 11: Stress waves propagating in Modes 1–4 in the four-mode corrected signal shown in Figure 10. Modes 2–4 are each offset by -100 MPa for clarity.

244 four-mode corrected signals have total impulses of 91.9 N s and 92.0 N s, respectively: both methods correct the  
 245 4% under-prediction from the recorded signal.

246 For applications requiring local accuracy of the signal, the correction is greatly improved by the inclusion of  
 247 the higher modes, which minimise spurious oscillations while resolving the shape and rise time of the main stress  
 248 wave. For example, the peak overpressure measured by the strain gauge is 275 MPa, and four-mode dispersion  
 249 correction calculates that this peak would be 289 MPa at the loaded face of the bar. The additional oscillations in  
 250 the one-mode signal result in a calculated peak overpressure of 320 MPa, over-predicting the four-mode value by  
 251 11%.

252 The contribution of each mode to the four-mode corrected signal is shown in Figure 11. Although the majority  
 253 of the signal propagates in Mode 1, the waves propagating in higher modes have amplitudes of up to 40 MPa,  
 254 illustrating why one-mode correction is inadequate in this example. The accurate temporal positioning of the  
 255 stress in higher modes is of particular importance when identifying key high-frequency features in blast loading  
 256 recordings, such as rise time, peak overpressure and wave reflections observed within the signal. Without higher-  
 257 mode correction, these features can appear at the wrong position in the signal, have the wrong magnitude, or be  
 258 lost altogether due to interference of the frequency components. In the example shown the oscillations following

259 the main wave are not entirely removed by either correction method, but this is likely to be due to the data lost on  
260 the arrival of the reflected wave after  $550 \mu\text{s}$ , and experiments could be designed to eliminate this issue.

## 261 **5. Summary and conclusions**

262 A pressure signal will disperse as it travels down a Hopkinson pressure bar because each component propagates  
263 at a velocity which is dependent on its frequency. Whilst methods exist for correcting this dispersion at lower  
264 frequencies, above a certain frequency a signal will propagate in multiple modes, at which point existing methods  
265 lose accuracy and validity. In this paper, we developed a method for correcting dispersion in higher modes based  
266 on a comprehensive numerical study using a validated finite element model.

267 First, we determined that finite element modelling of a Hopkinson pressure bar could provide quantitative  
268 data on the modal distribution of stress with frequency for use in higher-mode dispersion correction. Numerical  
269 simulations demonstrated that the dispersion and cross-sectional variation of both single-mode and multiple-mode  
270 stress waves in a cylindrical bar were accurately predicted, and so there was confidence that the distribution of  
271 stress between modes would also be modelled accurately.

272 Propagation in higher modes was assessed using short raised-cosine windowed pulses, which dispersed into  
273 multiple modal pulses as the signal propagated along the bar, enabling the contribution of each mode to be mea-  
274 sured up to a normalised frequency of 0.90. All four modes in this range were shown to contribute significantly at  
275 frequencies above their cutoff frequency, and dominance of a mode did not closely relate to group velocity.

276 Finally, we presented four-mode dispersion correction of an experimental signal. The results showed a marked  
277 improvement over the existing single-mode method, highlighting the importance of multi-mode dispersion correc-  
278 tion, as well as demonstrating the applicability and validity of finite element modelling as a tool for understanding  
279 multi-mode wave propagation behaviour. The dispersion correction algorithm allows for correction of the first  
280 four modes, up to normalised frequencies of  $fa/c_0 = 0.90$ , extending the accuracy and applicability of established  
281 frequency-domain techniques well beyond the Mode 2 cutoff frequency of  $fa/c_0 \approx 0.35$ .

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