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Meza, F.J., Becque, J. and Hajirasouliha, I. orcid.org/0000-0003-2597-8200 (2020) Experimental study of the cross-sectional capacity of cold-formed steel built-up columns. Thin-Walled Structures, 155. 106958. ISSN 0263-8231

https://doi.org/10.1016/j.tws.2020.106958

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# EXPERIMENTAL STUDY OF THE CROSS-SECTIONAL CAPACITY OF COLD-FORMED STEEL BUILT-UP COLUMNS

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6 Abstract: This paper describes a comprehensive experimental programme in which built-up cold-formed steel 7 stub columns with four different cross-sectional geometries were investigated. Twenty built-up sections were 8 fabricated from individual channels and flat plates with nominal thicknesses ranging from 1.2 mm to 2.4 mm 9 and assembled with either bolts or self-drilling screws. The connector spacing was varied among specimens of 10 the same geometry in order to study its effect. The built-up columns were tested between fixed boundary 11 conditions and the load was transmitted through end plates which were attached to the columns with an epoxy 12 resin. Tensile coupons were taken from the corners and flat portions of the constituent sections in order to 13 determine their material properties, while detailed measurements of the geometric imperfections of each 14 specimen were also performed using a laser displacement sensor. The experiments revealed a significant 15 amount of restraint and interaction between the individual components of the columns while buckling, with 16 the connector spacing having a pronounced effect on the observed buckling modes. However, the ultimate 17 cross-sectional capacity was seen to be much less dependent on the connector spacing within the considered 18 range of spacings.

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20 Keywords: Built-up column; Experiment; Cold-formed steel; Stability; Buckling; Imperfection
21 measurements.

#### 22 **1. Introduction**

23 Cold-formed steel (CFS) sections offer many important benefits to construction, such as high strength-to-24 weight and stiffness-to-weight ratios, an ease of handling, transportation and stacking, and important 25 sustainability credentials due to their recyclability and efficient material use. For these reasons their range of 26 application has rapidly expanded from being mainly used as secondary members in steel structures to an 27 increasing use as primary members. Examples are multi-storey buildings constructed entirely out of CFS [1] 28 and CFS portal frames [2]. This trend in construction is exerting an increased demand on CFS structural 29 members in terms of the span length and the load carrying capacity they need to provide. One way of meeting 30 these increased requirements is by joining two or more sections together by means of welding or fasteners 31 such as bolts, rivets or screws, to form a built-up section. A wider range of cross-sectional shapes can thus be 32 obtained using the currently available single shapes and cross-sections can be tailored to meet specific 33 requirements. In addition, double symmetry, which is difficult to obtain in single CFS sections, can easily be 34 achieved, while closed sections with increased torsional resistance can also readily be constructed. However, a 35 gap in understanding of the way these sections behave and a lack of specific design provisions in the current 36 design codes (e.g. AISI S100 [3]; EN 1993-1-3 [4]) has prevented the exploitation of their full potential. 37 Designing a built-up section requires careful consideration of various factors, including the type of connector 38 used, the spacing between connectors, the degree to which the individual components work as one cross-39 section, and the occurrence of individual as well as coupled instabilities.

40 CFS built-up members have already attracted significant research interest, especially during the past decade. 41 With regard to built-up compression members, most of the previous research has focused on either I-shaped 42 cross-sections constructed by connecting two identical channels or sigma sections through their webs in a back-to-back configuration, or built-up closed sections obtained by connecting channel sections through their 43 44 flanges. These types of cross-sections are commonly found in CFS roof trusses and framing systems used in 45 multi-storey buildings. Research on CFS I-shaped built-up columns includes the experimental work presented 46 by Stone and LaBoube [5], where the specimens were constructed from lipped channels connected with 47 screws. The specimens were seated in track sections at each end in order to replicate the usual end conditions 48 encountered in CFS framing systems. More recently, Fratamico et al. [6] also tested I-shaped columns seated 49 in tracks. However, the researchers added a group of connectors at each end of the column, as prescribed in 50 AISI S100 [3], to study its effect on the amount of composite action achieved in the columns. Lu et al. [7] also 51 conducted noteworthy research on screw-connected back-to-back lipped channel columns. In this latter 52 experimental programme endplates were welded to the column, avoiding any possible end slip between the 53 components. The authors observed that a significant amount of composite action could be achieved when the 54 failure mode was governed by global buckling, while composite action could be disregarded when the failure 55 mode of the tested columns was predominantly local or distortional buckling. Zhang and Young [8] conducted 56 compression tests on I-shaped built-up columns constructed from lipped sigma sections. The specimens were 57 assembled with self-tapping screws and were designed to exhibit local, distortional and global buckling, as 58 well as interaction between these buckling modes. The tests suggested that some degree of composite 59 behaviour was present. Abbasi et al. [9] further investigated this built-up geometry by carrying out a stability 60 analysis using the compound strip method to study the effect of the connector spacing on the buckling 61 behaviour of the columns, and showed that reducing the connector spacing enhanced the elastic local and 62 global buckling stresses of the built-up member.

63 An experimental investigation on screw-connected built-up columns with various geometries, including open 64 and closed sections, assembled from up to four individual channels, was conducted in [10]. The built-up 65 specimens were tested with pin-ended or fixed-ended boundary conditions and failed by interaction between 66 cross-sectional and global buckling. The study revealed a significant degree of composite action, with the 67 built-up geometries exhibiting ultimate capacities larger than those obtained by simply adding up the 68 contributions of the individual components. Liao et al. [11] carried out an experimental and numerical 69 investigation of multi-limb built-up stub columns with geometries similar to those tested in [10] and found 70 that the connector spacing had little impact on the ultimate capacity of the stub columns. Compression tests on 71 simply supported built-up closed sections assembled from lipped channels welded together at regular intervals 72 in a toe-to-toe configuration were carried out in [12]. The investigated cross-sections were relatively stocky 73 and the specimens failed due to global instabilities without the presence of local buckling. Reyes and Guzmán 74 [13] carried out experiments on built-up closed section columns of similar geometry, but employed channels 75 with more slender cross-sections. The specimens were tested between either fixed or flexible support 76 conditions, and were relatively short, failing predominantly due to local buckling. The tests showed no 77 statistical reduction in the ultimate capacity due to a larger connector spacing when failure was governed by 78 local buckling. Young and Chen [14] and Zhang and Young [15] experimentally studied screw-connected 79 built-up closed sections assembled from two identical sigma sections or channel sections, respectively. The 80 columns were tested between fixed end conditions, with endplates welded to each end, and their failure modes 81 involved local, distortional and global flexural buckling. The tests showed that the ultimate capacity of the 82 built-up members was generally larger than that obtained by simply adding the contributions of the individual 83 channels. Li et al. [16] also carried out compression tests on built-up closed section columns assembled from 84 lipped channels with intermediate web stiffeners. However, the specimens were tested without welded 85 endplates, resulting in some of the specimens failing prematurely due to an uneven distribution of the load 86 between the components. Georgieva et al. [17-20] carried out an extensive experimental and numerical 87 investigation to study the buckling behaviour of double-Z built-up columns. The columns were assembled 88 with spacers bolted to both Z-sections to restrain distortion of the cross-section at discrete points and were 89 designed to exhibit local, distortional and global buckling.

A simple design equation was proposed in [21] to predict the ultimate capacity of I-shaped and closed section built-up columns assembled from lipped channels stitch-welded together at various spacings and failing by distortional or global flexural buckling. The equation was based on the experimental and numerical data gathered in [12,22,23]. However, the authors suggested that more compression tests on other practical built-up geometries were needed to further validate the proposed equation.

95 Despite the significant amount of research focused on built-up compression members to date, most of the 96 previously investigated geometries were fabricated from two identical components, with both components 97 buckling at the same time. In contrast to this, the experimental programme described in this paper consisted of 98 20 stub column tests of four different built-up geometries, each constructed from four individual components 99 including two different cross-sections. The cross-sectional components were brake-pressed from galvanized 100 steel sheet with thicknesses between 1.2 mm and 2.4 mm and were connected with either M6 bolts or M5.5 101 self-drilling screws. The aim of the study was to investigate the cross-sectional compressive behaviour and 102 capacity of the tested geometries and quantify the influence of parameters specific to built-up sections. Most importantly, the effect of the longitudinal connector spacing on the cross-sectional stability and the ultimate capacity of the built-up specimens was investigated. Direct observations from the tests, as well as comparisons of the measured buckling loads with those calculated for isolated components, also provided information about the mutual restraint components offered to each other in the built-up configuration through contact and discrete connections. The targeted instabilities were local buckling of the components, as well as (overall) buckling of the components in between connection points. Previous related research by the authors has also covered the behaviour and capacity of CFS built-up beams [24] and CFS built-up long columns [25].

#### 110 **2. Section geometry and specimen fabrication**

111 Figure 1 illustrates the four cross-sectional geometries included in the tests. The individual components were 112 brake-pressed by an external fabricator and were given a specific label. Plain channels, lipped channels and 113 flat plate components were labelled using the letters 'T', 'S' or 'P', respectively, followed by the nominal 114 height of their web in mm (in the case of the channels) or the nominal width of the plate in mm, and the 115 nominal wall thickness in mm multiplied by 10. To refer to each built-up specimen, the letters 'SC' were used 116 to indicate that the specimen was a stub column, followed by a number ranging from 1 to 4 to indicate its 117 cross-sectional geometry (with reference to Figure 1), a hyphen and the number of rows of intermediate 118 connectors (i.e. not counting the connectors in the end sections). As each test was repeated, the letters 'a' and 119 'b' were used to indicate whether the specimen was the first or the second of twin columns tested. For 120 example, the label 'SC1-2a' referred to the first stub column tested with cross-section geometry 1 which 121 contained two intermediate rows of connectors between the end connectors.

122 Sections P20024, T15414 and T7912 (i.e. those used to assemble columns SC1 and SC2) were manufactured 123 from pre-galvanized steel sheets with a guaranteed yield stress of 450 MPa and a Z275 zinc coating with a 124 total nominal thickness of 0.04 mm. Sections T12012 and S11012 (i.e. those used to assemble columns SC3 125 and SC4) had a guaranteed yield stress of 260 MPa, and although for these sections the coating thickness was 126 not specified by the manufacturer, it was determined to be 0.03 mm by measuring the thickness of the sections 127 at each end before and after removing the zinc coating with hydrochloric acid. The thickness of the sections 128 was measured using a digital micrometre with a precision of  $\pm 0.002$  mm, while the rest of the dimensional 129 measurements were carried out using a digital Vernier caliper with a precision of ±0.03 mm. All measurements were of the outside dimensions, as illustrated in Figure 2. Table 1 lists the nominal dimensions of the components (using the symbols introduced in Figure 2), while Tables 2-5 summarize the measured dimensions for columns SC1-SC4, respectively. Each listed value is the average result of several measurements taken at different locations. The thickness values correspond to the base metal and were obtained after deducting the thickness of the zinc coating.

135 All stub columns were designed to fail by cross-sectional instability, buckling of the individual components 136 between connectors, or a combination of both, but without influence of global instabilities. In this respect the 137 recommendations of the 'Column Research Council' were followed, which state that the length of the column 138 should be larger than three times the largest dimension of the cross-section and smaller than 20 times the least 139 radius of gyration [26]. As a result, the length chosen for columns SC1, SC3 and SC4 was 1100 mm and the 140 length of columns SC2 was 800 mm. At these lengths, all columns were expected to buckle with at least six 141 half-waves along their length. This resulted in at least one buckling wave (in the middle) where the influence 142 of the boundary conditions was minimal. An estimate of the local bucking load was obtained by carrying out a 143 finite strip analysis of the column components, considered in isolation without contact or connections with the 144 other components, using CUFSM [27]. Flexural buckling of the flat plates between connector points in 145 columns SC1 was predicted using the classical (elastic) equation, using a buckling length equal to half the 146 connector spacing.

147 The connector spacing was varied among specimens of the same geometry in order to study its influence. 148 Columns SC1 were designed with 2, 3 and 5 equally spaced intermediate rows of connectors, corresponding to 149 configurations where the predicted flexural buckling stress of the flat plates was lower than, equal to and 150 higher than the predicted local buckling stress of the channels, respectively. Columns SC2 were designed with 151 2, 4 and 6 rows, and columns SC3 and SC4 were designed with 2 and 5 rows, as shown in Figure 3. These 152 connector spacings were chosen based on the predicted local buckling half-wavelengths of the components. In 153 columns SC2-6 the connector spacing was shorter than twice the natural half-wavelength of the channels, 154 while in columns SC2-2 the connector spacing was larger. In columns SC3-5 and SC4-5 the connector spacing 155 coincided with twice the natural local buckling half-wavelength of the lipped channels, but was shorter than 156 twice the half-wavelength of the plain channels, while in columns SC3-2 and SC4-2 the connector spacing 157 was larger than twice the natural half-wavelength of all channels.

The test specimens were constructed using two different types of connectors, to study whether this influenced their behaviour. Columns SC1 and SC2 were assembled using M6 bolts, while columns SC3 and SC4 were assembled using M5.5 self-drilling sheet metal screws (Figure 4). For more detailed considerations regarding the design of the test specimens the reader is referred to [28].

162 In order to assemble columns SC1 and SC2, bolt holes with a diameter of 6.25 mm were drilled in the 163 appropriate locations. In columns SC1, the holes were first drilled in the flat plate components at the locations 164 shown in Figure 3a. For columns SC2, the holes were first drilled in the outer channels at the locations 165 illustrated in Figure 3b. These sections were then used as templates to drill the holes in the remaining sections 166 after positioning all sections in their built-up configuration and securing them with clamps. The components 167 were then bolted together using a torque wrench, while applying a torque of 10 Nm. This value was judged to 168 be representative of manually tightening the bolts with a spanner. A similar procedure was followed to 169 assemble columns SC3 and SC4. The locations of the screws were first marked in one of the connecting 170 components and small diameter holes were drilled in order to facilitate the installation of the screws. Next, the 171 sections were positioned in their built-up configuration, secured with clamps and screwed together. The 172 locations of the screws are illustrated in Figure 3c and Error! Reference source not found. Figure 3d for 173 columns SC3 and SC4, respectively. Figures 5a-c show the finalized columns SC1-SC4.

174 After the assembly process, both ends of each column were manually filed to correct any differences in length 175 between the individual components. Much care was put into this process to ensure that a completely even 176 bearing surface was obtained between the specimen ends and the end plates. Endplates with dimensions of 177  $250 \times 300 \text{ mm}^2$  and a thickness of 20 mm were then attached to both ends of columns SC1 and SC2, while 178 endplates with dimensions of 200×200 mm<sup>2</sup> and a thickness of 20 mm were attached to the ends of columns 179 SC3 and SC4. It was decided not to weld the endplates to the columns because of the limited wall thickness of 180 the sections and the concern that the welding process would introduce considerable distortions into the 181 sections. Instead, a Sikadur 31 FC Normal 2-part thixotropic epoxy resin was used to attach the endplates. The 182 zinc coating was removed over a distance of 30 mm at the column ends to improve bond and the resin was poured into a mould around the column base up to a height of 20 mm. The epoxy resin was specified by the manufacturer to have a bond strength of 18 MPa and an elastic modulus of 6.6 GPa. Consequently, the height of 20 mm was sufficient to transfer a load of 350 kN in shear through the interface with the steel column, which well exceeded the column capacity for each specimen. In addition to providing a double row of connectors at each column end and manually filing the ends, the resin thus provided an extra means to ensure a uniform introduction of the load into all components.

#### 189 **3. Material Properties**

190 The material properties of the test specimens were obtained by carrying out a series of tensile coupon tests. 191 The coupons were cut from spare sections belonging to the same batch as those used in the test. Two flat 192 coupons were taken along the centre line of the web of each type of channel section and along the centre line 193 of the flat plate used in columns SC1. In addition, two corner coupons were cut from the web-flange junction 194 of each type of channel section. Therefore, 18 coupons were tested in total. The dimensions of all coupons 195 adhered to the specifications provided in [29]. In particular, all flat coupons had a nominal width of 12.5 mm 196 and a gauge length of 50 mm. Each coupon was instrumented with an extensioneter and one linear 5 mm 197 strain gauge on each side of the coupon. After removing the zinc coating with an emery cloth, the width and 198 thickness of each coupon were measured using a digital Vernier caliper with a precision of  $\pm 0.03$  mm. All 199 corner coupons had a nominal width of 6 mm. The coupons taken from the components of columns SC1 and 200 SC2 had a gauge length of 25 mm. However, this gauge length was found to be too short to allow for an easy 201 installation of the instrumentation onto the coupon, and therefore, the corner coupons taken from the 202 components of columns SC3 and SC4 were subsequently cut with a gauge length of 50 mm. The corner 203 coupons were tested in pairs in order to avoid eccentric tension. Two coupons extracted from the same type of 204 channel were tested together, with a square bar placed in between the gripped ends, as shown in Figure 6. 205 Each pair of corner coupons was instrumented with an extensioneter. In addition, a 5 mm strain gauge was 206 fitted to the outside of each coupon.

The cross-sectional area of the corner coupons was determined by photographing the end sections of the coupon using the reverse lens technique (Figure 7), importing the image into AutoCAD® software, superimposing the width of the coupon along the gauge length onto the picture and using the software features to automatically calculate the area of interest. The process was repeated with pictures taken from the oppositeend of the coupon and agreement within 2-3% was obtained in all cases.

All coupons were tested following the specifications provided in the relevant European standard [29]. The tests were performed in a 300 kN Shimadzu AGS-X universal testing machine with a displacement rate of 1-2 mm/min. Each test was halted for 2 minutes at regular intervals, with the first pause imposed when yielding was first observed in the coupon. This allowed the load to settle down to 'static' values and eliminated strain rate effects. As an example, Figure 8 shows the ('static' and 'dynamic') stress-strain curves of one of the flat coupons and the pair of corner coupons taken from channel T12012. Stresses and strains shown in this figure are conventional 'engineering' values.

219 Table 6 lists the (static engineering) values of the 0.2 % proof stress ( $\sigma_{0.2\%}$ ), the ultimate tensile strength ( $\sigma_u$ ) 220 and the elongation at fracture ( $\varepsilon_f$ ) obtained for each coupon, as well as the average values over twin coupons. 221 For the corner coupons extracted from sections T15414 and T7912 the elongation at fracture was based on a 222 gauge length of 25 mm, while for the rest of corner coupons and all of the flat coupons this was based on a 223 gauge length of 50 mm. It is important to note that although the 0.2 % proof stresses listed in the table are in 224 some cases lower than the nominally specified values (i.e. 450 MPa for sections P20024, T15414 and T7912, 225 and 260 MPa for sections T12012 and S11012), the values listed in the table correspond to the 'static' 0.2 % 226 proof stresses, reduced to zero strain rate. The nominal values reported in practice are based on 'dynamic' 227 values (with strain rates within the limits set by the standards).

#### **4. Imperfection Measurements**

Imperfections may have a significant influence on the stability of thin-walled structural members, particularly when coupled instabilities are involved [30,31] and the topic of imperfections in cold-formed steel structural members has previously received considerable investigative effort [32, 33]. For this reason, the imperfections of all test specimens were recorded before testing. The equipment used to carry out the measurements consisted of a steel table with a very high degree of flatness, a traverse system powered by electric motors travelling at a pre-determined constant speed and a laser displacement sensor. The flat table with dimensions of 1500×920 mm<sup>2</sup> was made of cast iron and was classified to be grade 3 according to [34], meaning that it 236 provided a surface with a deviation from flatness of less than 0.06 mm. A traverse system consisting of an aluminium frame with a trolley, high-precision guiding bars and two electric motors was placed on top of the 237 238 flat table, as illustrated in Figure 9. The frame had dimensions of 2400×600 mm<sup>2</sup> and rested on four adjustable 239 supports. The two electric motors allowed movement of the laser sensor, which was attached to the trolley, 240 along two orthogonal horizontal axes. Movement in the vertical direction was controlled manually by turning 241 a crank handle located on the trolley (Figure 9). This permitted the laser sensor to be positioned within 242 measuring range from the surface. The laser displacement sensor was a Keyence LK-G82 sensor with a beam 243 spot diameter of 70  $\mu$ m, a measurement range between 65 and 95 mm and an accuracy of ±0.0075 mm.

244 The imperfections were measured by moving the laser displacement sensor along different longitudinal lines 245 on each face of the built-up column, as indicated by the red arrows in Figure 10 for each built-up geometry. 246 The black arrows indicate measurements taken of the individual components before assembly, as access to 247 them was restricted in the final configuration. Imperfection readings were considered positive in the direction 248 away from the centroid of the column. For columns SC1 and SC2 the laser sensor was moved at a constant 249 speed of 5 mm/s, while readings were taken with a sampling rate of 50 Hz, resulting in a reading every 0.1 250 mm. It was subsequently concluded that taking readings at such short intervals was not strictly necessary to 251 obtain a representative imperfection profile. Therefore, the sampling rate was reduced to 5 Hz for columns 252 SC3 and SC4, resulting in readings every 1 mm. Measurements of the nominally flat table, without a test 253 specimen present, were used to correct for the out-of-straightness of the guiding bars along which the laser 254 sensor was moved. Therefore, the accuracy of the measurements was determined by the flatness of the table 255 and of the order of 0.06 mm.

The out-of-plane imperfection measurements were used to determine representative imperfections of the column components. For both the plain and the lipped channels, the imperfections of interest included the outof-plane imperfections along the centre line of the web measured relative to the line connecting the corners  $(\delta_{web})$  and the out-of-plane imperfections along the flange edge measured relative to the web-flange junction  $\delta_{flange}$  (where the 'flange edge' either indicates the free edge in the case of a plain channel or the flange-lip junction in the case of a lipped channel). For the lipped channels the out-of-plane imperfections along the centre line of the flange measured relative to the corners ( $\delta_{flanges,L}$ ) were also considered. The flat plate components in columns SC1 were expected to buckle in a global flexural mode between connectors and for these sections the imperfections of interest ( $\delta_{plate}$ ) were computed as the average over the three measurement lines.

266 It is worth noting that the imperfections of the flanges of the channels in columns SC1 and the flanges of the 267 inner channels in columns SC2 were not measured, since there was not enough space within the channels to 268 place the laser sensor at an appropriate distance from the flanges. However, local buckling of these channel 269 sections was expected to be mainly affected by the imperfections in the web, as this constituted the most 270 slender part of the cross-section. Similarly, the imperfections of the lipped channels in columns SC3 and the 271 imperfections of the plain channels in columns SC4 could not be measured after the specimens were 272 assembled. Therefore, they were recorded prior to assembly. However, only the web imperfections of these 273 channels were considered to be important due to the slenderness of the web and the geometric arrangement of 274 the cross-section, where the flanges are connected to other components.

Figure 11 shows the out-of-plane imperfections recorded on a representative column with geometry SC1. The vertical dashed black lines indicate the locations of the connectors. The complete imperfection data of all specimens can be found in [28].

278 Table 7 lists, for each built-up geometry, the maximum and the average out-of-plane imperfections recorded 279 on the individual components. The average value of  $\delta_{flange}$  was computed assuming that  $\delta_{flange} = 0$  at the 280 column ends. In other words, it was assumed that the flanges were perfectly orthogonal to the web at the 281 column ends. However, to determine a maximum value of this imperfection it was deemed more 282 representative to report the maximum value of  $\delta_{flange}$  relative to the average value along the flange. The table 283 shows that the maximum recorded imperfections were generally smaller than 1 mm in all components. Only 284 channel T12012 in columns SC3 showed a maximum out-of-plane imperfection in the web larger than 1 mm. 285 However, this relatively large imperfection was only recorded in one channel. The rest of the T12012 channels 286 had maximum imperfections smaller than 0.57 mm. The recorded average imperfection was smaller than 0.36 287 mm in all components. It is worth pointing out that the maximum and average imperfections  $\delta_{flange,L}$  in the 288 S11012 lipped channels were smaller than the accuracy of the measuring frame and those values should 289 therefore only be taken as an indication that the imperfections were very small.

#### **5. Test Set up**

All specimens were tested between fixed end supports in an ESH universal testing machine with 1000 kN capacity. To monitor whether the load was uniformly transmitted to each component of the column, a total of four columns –one of each geometry– were instrumented with strain gauges at mid-height. One strain gauge was placed on each individual component of the built-up column, along the centre line of the web of the channel or along the centre line of the flat plate. For each geometry, one of the columns with the least amount of connectors was chosen to be instrumented since the effects of uneven participation of the various components would have been most pronounced in these columns.

298 Two vertical potentiometers were placed underneath the end plate at the top of the specimen, one on each side 299 of the column, to record the axial shortening of the specimens. The readings obtained from these two 300 potentiometers were in close agreement for all specimens, indicating that fixed support conditions were 301 successfully achieved and that no end rotation took place during the test. In addition, a number of 302 potentiometers were placed in a horizontal position, typically divided over two cross-sections, to measure the 303 out-of-plane deformations of the components and capture the onset of local buckling. Figure 12 illustrates the 304 typical arrangement used for the columns of different geometry, indicating both the measuring locations 305 within the cross-section and the vertical placement over the height of the specimens.

A consistent strain rate of  $1.7 \times 10^{-6}$  /s was applied to all specimens. This corresponded to a displacement rate of 0.112 mm/min for columns SC1, SC3 and SC4 (with a length of 1100 mm), and 0.082 mm/min for columns SC2 (with a length of 800 mm). Columns SC1 and SC2 were compressed over a range of 10 mm during the test, while columns SC3 and SC4 were compressed by up to 3.5 mm. This was sufficient to obtain the loaddisplacement graph until well past the peak load. Each test was halted for 3 minutes when approaching the peak load in order to determine the lower bound 'static' value of the load, independent of strain rate dependent effects.

<sup>313</sup> The data acquisition system was controlled by a LabView script, which imposed a sampling rate of 1 Hz.

#### 314 **6. Test results**

#### 315 **6.1. Strain gauge readings**

316 Figure 13 shows a typical set of strain gauge readings over the course of the test for a column with geometry 317 SC1. Compressive strains were taken as positive. The figure shows that the strains in the channel sections and 318 the flat plate sections were in good agreement until buckling occurred at a load of approximately 60 kN. 319 Below this load, the strain in the flat plate sections differed by at most 12 % from the column average, while 320 this number was 11 % for the channel sections. The readings also show that the plates were typically subject 321 to slightly lower strains than the channels. This can likely be attributed to the initial imperfections present in 322 the plate sections, as well as to their low flexural stiffness. These imperfections caused out-of-plane 323 deformations from an early load of approximately 10 kN, as evidenced by the potentiometer readings. Due to 324 the presence of the channels the plates could only bend outwards, which introduced superimposed tensile 325 stresses at the location of the strain gauges.

Similar general conclusions could be drawn for the other three geometries. Below the critical buckling load the load was observed to be evenly distributed over all components, with the maximum recorded difference in strain in the various components with respect to the column average ranging from 3% (geometry SC4) to 12% (geometry SC2). For geometry SC3 this value was 5%.

#### **6.2. Observations of the deformations and failure modes**

#### **6.2.1 Columns SC1**

All columns SC1 failed by interaction of global buckling of the flat plate components between connectors and local buckling of the channels. Each pair of twin columns showed the same initial buckled shape (Figure 14), although the eventual localization of the buckling pattern with the formation of a yield line pattern often occurred in different locations.

In columns SC1-2, which exhibited the largest connector spacing, the flat plate components buckled outwards in between connector points in a global-type flexural mode with a half-wavelength slightly beyond half the connector spacing (Figure 14a). However, the observation that the buckling deformations of the flat plates

339 were localized in the central field of the column, with the adjacent fields remaining straight, was 340 unanticipated. Seen the regularity of the buckling pattern in the adjacent channels, it is clear that a certain 341 amount of bolt slip necessarily occurred in order to make this possible. The channel components maintained 342 deformational compatibility with the unbuckled plate sections in the end fields by the flanges buckling 343 inwards over the whole length of the column. The channels buckled with a half-wavelength equal to half the 344 connector spacing. This can be explained by the fact that the signature curve of the unrestrained channels 345 (Figure 15), obtained using CUFSM [27], displayed a fairly flat local buckling minimum for half-wavelengths 346 between 140-200 mm, while the connector spacing of columns SC1-2 was 333 mm. The buckle half-347 wavelength of the flanges was slightly adjusted to be compatible with the buckle half-wavelength of the 348 plates. In the post-peak range the channel deformations localized in the central field, while the buckling 349 pattern largely disappeared in the end fields.

In columns SC1-3 the channel components buckled in a similar way, with the flanges bending inwards. The half-wavelength extended slightly beyond half the connector spacing, a fact which was particularly evident in the two central fields of the column. The flat plate components buckled outwards in two of the fields, with a half-wavelength sympathetic to the one observed in the channels, while they remained largely straight in the other two fields, prevented from buckling inwards by the presence of the channel webs. Similarly to columns SC1-2, the post-peak deformations localized in a field where the plates buckled outwards.

356 Columns SC1-5 featured a connector spacing of 167 mm, which was approximately equal to the natural local 357 buckle half-wavelength of the channel components (Figure 15). Unlike what was observed in columns SC1-2 358 and SC1-3, the channels in columns SC1-5 buckled with a half-wavelength equal to the distance between the 359 connectors, with the flanges alternatingly moving inwards and outwards in successive fields. The connector 360 locations corresponded to the nodal lines of the buckling pattern, implying that they did not undergo any out-361 of-plane translations, but did accommodate the plate rotations. The flat plate components buckled in a shape 362 which followed the flange tips in the fields where these moved outwards, while they remained straight in the 363 fields where the flange tips moved inwards, prevented from maintaining complete conformity by the presence 364 of the channel webs. This buckling pattern required some concentrated bending in the plate components 365 around the connectors. Post-peak localization of the buckled shape occurred in the central field, where the 366 plate components initially moved outwards.

367 It can be concluded from the above that the buckling pattern in columns SC1 was highly dependent on the 368 connector spacing.

#### 369 **6.2.2 Columns SC2**

Unlike in the columns SC1, where global buckling of one of the components in between connector points was prevalent, the buckling modes of the components of columns SC2 were all local. Columns SC2 aimed to study the interaction between the local buckling patterns in the inner and outer channels as a result of the presence of the connectors and contact between surfaces. Figure 16 illustrates the deformed shapes of the columns just before the peak load was reached.

375 In the columns with two rows of intermediate connectors (SC2-2) the buckle half-wavelength of the outer 376 channels coincided with half the distance between the connectors. Due to the presence of the inner channels, 377 the webs of the outer channels were forced to buckle outwards (away from the centre of the column). The 378 buckled shape is shown in Figure 16a. It is noted that the natural half-wavelength of the outer channels in 379 isolation was calculated to be 170 mm (Figure 17) using CUFSM [27], while the connector spacing was 233 380 mm. This stability analysis also revealed that for the outer channels, generating two half-waves between 381 connectors is associated with a buckling stress which is lower than for any other pattern with an integer 382 number of half-waves between connectors.

In columns SC2-4 the outer channels buckled with a half-wavelength equal to the distance between the connectors. This buckling pattern is shown in Figure 16b and can also be explained based on the results of the CUFSM buckling analysis of the unrestrained channel (Figure 17). The critical stress associated with a halfwavelength equal to the distance between connectors ( $L_{cr} = 140$  mm) is 65 MPa, whereas generating buckles half that length ( $L_{cr} = 70$  mm) would require a stress level of 112 MPa.

In columns SC2-6 the outer channels were observed to buckle with varying half-wavelength along the column. Along part of the specimen height the outer channels buckled with a half-wavelength very close to, but slightly beyond, the distance between connectors. However, this pattern switched to one in which the outer 391 channels displayed a half-wave which spanned almost two fields over an intermediate connector, as illustrated 392 in Figure 16c. This can be explained by the fact that these observed half-wavelengths are associated with a 393 lower critical stress in the signature diagram than half-wavelengths equal to the connector spacing of 100 mm 394 or twice that distance.

In all columns SC2 the inner channels buckled with a half-wavelength equal to half the distance between the connectors. Due to the presence of the outer channel webs, the flanges of the inner channels were forced to buckle towards the inside of the channels. This pattern occurred in all columns, despite the wide range of connector spacings, indicating a high degree of restraint on the inner channels.

### 399 **6.2.3 Columns SC3**

All columns SC3 failed by local buckling. Figure 18 shows the deformed shapes of the columns right before
the peak load was reached, while a representative failure mechanism is shown in Figure 19 for column SC35b.

In all columns the lipped inner channels buckled with a half-wavelength of approximately 83 mm, corresponding to four and two half-waves between connectors for columns SC3-2 and SC3-5, respectively. In comparison, the natural local buckle half-wavelength of the unrestrained lipped channel is 90 mm, as shown by the signature diagram in Figure 20.

407 In columns SC3-2 the plain outer channels buckled with two to four half-waves between connectors. The 408 cross-sections containing connectors thereby corresponded to the minima of the buckling pattern. In column 409 SC3-2a the flanges of the outer channels followed the out-of-plane displacements of the webs of the inner 410 channels almost perfectly. In its twin specimen SC3-2b, however, this was not the case, causing a more 411 pronounced gap between the inner and outer channels, as is visible in Figure 18b. In this respect, it is also 412 worth noting that the natural local buckle half-wavelength of the plain channels was 130 mm, which was 413 associated with a critical stress of 79 MPa. Half-wave lengths of 83 mm (a quarter of the connector spacing) 414 and 167 mm (half of the connector spacing) corresponded to buckling stresses of 91 MPa and 83 MPa, 415 respectively.

416 A more regular buckling pattern was observed in the plain channels of the columns with five rows of 417 intermediate connectors. In these columns, the web of the plain channels buckled with a half-wave length 418 equal to half the distance between connectors (83 mm). This occurred despite the slightly larger buckling 419 stress associated with this wavelength in comparison to a wavelength twice as long, and is indicative of the 420 amount of restraint exerted by the inner onto the outer channels. In column SC3-5b the plain and lipped 421 channels buckled in near complete sympathy, while in column SC3-5a the flanges of the plain channels (being 422 less restrained by the connectors than the web) displayed a half-wavelength closer to their natural half-423 wavelength, resulting in the formation of some gaps between them and the webs of the lipped channels 424 (Figure 18c).

### 425 6.2.4 Columns SC4

Columns SC4 failed predominantly by local buckling, with some minor participation of distortional buckling of the lipped channels. The potentiometer readings indicated that in columns SC4-2a, SC4-2b and SC4-5a, a minor amount of distortional buckling originated in one of the lipped channels just before the peak load was reached, while in column SC4-5b a minor amount of distortional buckling developed in both lipped channels simultaneously. Figure 21 illustrates the deformed shape of the columns shortly before the peak load was reached, while Figure 22 shows an example of the localized failure mechanism past the ultimate load.

In all SC4 columns the lipped channels first buckled in a local mode with a half-wavelength of approximately 83 mm, corresponding to a quarter of the connector spacing and half of the connector spacing in specimens SC4-2 and SC4-5, respectively. As the webs of the plain channels prevented the webs of the lipped channels from buckling towards the inside of the column, the flanges of the lipped channels were forced to buckle inwards.

#### 437 **6.3. Critical buckling stresses**

An estimate of the buckling stress of the critical components of the built-up specimens was derived from the potentiometers readings by assuming that, up to the point of first buckling, the load was uniformly distributed over all components, and that the initial post-buckling out of plane displacements follow a parabolic trend, as shown in Figure 23 for column SC2-2a. After fitting a parabola to these initial displacements, the buckling 442 load followed from the intersection of this parabola with the vertical axis. Further details and examples, as 443 well as all potentiometer readings as a function of the load applied to the column, can be found in [28]. For 444 those columns instrumented with strain gauges these buckling stresses were in good agreement with the values 445 obtained from the strain gauge readings using the strain reversal method [35].

446 The experimental buckling stresses were compared to theoretical buckling stresses, calculated based on the 447 measured cross-sectional dimensions (averaged over the two components in the cross-section) and using the 448 Young's modulus obtained from the flat tensile coupons (Tables 8-11). The theoretical buckling stresses were 449 obtained while considering the individual components in isolation, without any interaction with the rest of the 450 cross-section, and adopting the buckle half-wavelength observed during the test. For the plain and lipped 451 channel sections the theoretical critical buckling stresses were obtained using the CUFSM 4.05 software [27], 452 while for the plate components in columns SC1 the critical buckling stress was determined using Euler's 453 equation:

$$\sigma_{cr} = \frac{\pi^2 E t^2}{12L_p^2} \tag{1}$$

In the above Eq. (1), *E* is the Young's modulus, *t* is the averaged measured thickness of the two plate sections in the column and  $L_p$  is the buckle half-wave length (effective length). Eq. (1) was evaluated using two different values of  $L_p$ . An upper bound for  $\sigma_{cr}$  was established by taking  $L_p$  equal to half the connector spacing, while a lower bound was obtained by assuming  $L_p$  to be equal to the connector spacing (Table 8).

458 Table 8 shows that the experimental buckling stresses of the flat plate components were intermediate between 459 the theoretical lower and upper bounds. This agrees with the experimentally observed buckled shape, as 460 described in Section 6.2.1. An estimate of the local buckling stress of the channels in the SC1 columns could 461 also be derived from the experiment by assuming that, due to the lack of post-buckling capacity in the global 462 flexural mode, any increase in load after buckling of the flat plate took place was entirely resisted by the 463 channels. This approach demonstrated that the channels buckled at a stress very close to the theoretically 464 predicted value. This suggests that the channels were not significantly affected by any restraint provided by 465 the plates, which buckled either before or simultaneously with the channels.

466 Table 9 lists the theoretically predicted buckling stresses of the inner and outer channels of the SC2 columns. 467 Only the buckling stresses of the outer channels could be experimentally determined (by dividing the load at 468 which their buckling was observed over the gross cross-sectional area), since the load sharing between the 469 outer channels in their post-buckled state and the inner channels is a non-trivial problem. In some columns the 470 two outer channels buckled at slightly different loads, as shown in Figure 23, and in this case both values are 471 reported in Table 9. The buckling stresses of the outer channels obtained from the test were generally slightly 472 larger than the predicted values, with the difference increasing as the connector spacing was reduced. For 473 columns SC2-6 the difference became quite substantial, with the buckling stresses obtained from the test being 474 around 35 % larger than the theoretical ones. This indicates a more significant amount of restraint exerted by 475 the inner channels onto the outer channels as the connector spacing decreased.

476 The theoretical and experimental critical buckling stresses of the components of the SC3 columns are listed in 477 Table 10. The plain and the lipped channels buckled approximately at the same time in all columns. In those 478 columns in which the plain channels were seen to buckle in a mixed pattern with two different buckle half-479 wavelengths, the stresses associated with both half-wavelengths are included in the table. The channels were 480 experimentally observed to buckle at a stress level of approximately 103 MPa. This stress is very close to the 481 theoretically predicted buckling stress of the lipped channels, while it is around 40 % higher than the buckling 482 stress predicted for the plain channels. This indicates that the plain channels substantially benefited from the 483 restraint provided by the lipped channels.

484 Table 11 lists the theoretical and experimental buckling stresses of the different components of the SC4 485 columns. In columns SC4-2, the plain channels buckled before the lipped channels, while in columns SC4-5 486 all components buckled at approximately the same time. This allowed the experimental buckling stresses of 487 the lipped channels of columns SC4-5, as well as the experimental buckling stresses of the plain channels of 488 all columns to be determined. Table 11 shows that the plain channels buckled at an increased stress level 489 compared to the theoretical predictions. This was due to the restraint exerted by the lipped channels, which 490 forced the flanges of the plain channels to buckle inwards. The lipped channels, on the other hand, buckled at 491 stresses close to, or slightly below, the theoretically predicted values.

#### 492 **6.4. Ultimate load**

Figure 24a-d plot the (static) load vs. axial displacement curves of all columns with geometries SC1, SC2, SC3 and SC4, respectively. It is worth pointing out the marked decrease in the stiffness of columns SC1 after first buckling of the plate components took place, compared to the more gradual decrease in stiffness observed in the columns with geometries SC2, SC3 and SC4. This can be explained by the fact that the plates contributed 57 % to the total cross-sectional area of the SC1 columns, while after buckling in a global flexural mode, these plates were unable to contribute in resisting a further increase in load.

The ultimate loads obtained for all columns with geometries SC1, SC2, SC3 and SC4 are listed in Table 12.
The average value for each set of twin columns is also provided.

Table 12 shows that, regarding columns SC1, the difference in the ultimate load achieved in twin specimens was 9 % for columns SC1-2, 4 % for columns SC1-3 and 7 % for columns SC1-5. The tests also showed a moderate increase in ultimate load as the spacing between connectors was reduced. More specifically, halving the connector spacing from 333 mm to 167 mm produced an increase in the average ultimate load of 11 %.

In columns SC2 good agreement was again obtained between the results of twin specimens. The difference in ultimate load was 6 % for columns SC2-2, 2 % for columns SC2-4 and 5 % for columns SC2-6. In this case the tests showed that reducing the spacing between connectors did not necessarily result in a noticeable increase in ultimate load. The columns with a connector spacing of 140 mm (columns SC2-4) showed similar ultimate loads compared to the columns with connectors spaced every 100 mm (columns SC2-6). However, the largest connector spacing (233 mm) did result in a slightly lower ultimate load, which was on average 9 % below that of the SC2-6 columns.

The difference in ultimate load between twin specimens of the SC3 columns was 3 % for columns SC3-5 and just 1 % for columns SC3-2. The results also show that halving the connector spacing from 333 mm to 167 mm only resulted in a negligible increase in ultimate capacity of 2 %.

With respect to columns SC4, the difference in ultimate load between twin specimens was 3 % for columns SC4-5 and 0.7 % for columns SC4-2. In this case, reducing the connector spacing actually resulted in a slight reduction in the ultimate capacity. More specifically, halving the connector spacing from 333 mm to 167 mm 518 caused a reduction in the ultimate capacity of 6 %. This difference is quite marginal and might be due to the 519 statistical variation of the relevant parameters (imperfections, geometry, material properties, etc.).

## 520 7. Conclusions

521 An experimental program was carried out consisting of 20 built-up thin-walled stub columns with four 522 different cross-sectional geometries. The cross-sections were assembled from flat plates, plain channels and 523 lipped channels with nominal thicknesses ranging from 1.2 mm to 2.4 mm. Two of the cross-sectional 524 geometries (SC1 and SC2) were assembled using M6 bolts, while the other two (SC3 and SC4) were assembled using M5.5 self-drilling sheet metal screws. The connector spacing was varied among specimens of 525 526 the same cross-sectional geometry. The experimental investigation included tensile coupon tests to determine 527 the material properties of the flat portions and the corner regions of the different components. Accurate 528 measurements of the out-of-plane geometric imperfections of the specimens were also carried out using a laser 529 sensor. The columns were compressed between fixed supports in a displacement controlled regime. Strain 530 gauge readings obtained from a select number of specimens confirmed a uniform introduction of the load into 531 all components of the cross-sections. This was achieved thanks to hand-filing the end sections, placing a 532 double row of connectors at the specimen ends and epoxy-gluing the end plates to the specimens.

Substantially different buckling patterns were generally observed in columns with the same cross-sectional geometry, but different connector spacing. The observations of the deformed column shapes also indicated that the buckling patterns of the individual components within the columns were subject to considerable restraint. This restraint manifested itself in two different ways: (1) a change in the natural local buckle halfwavelength to accommodate the presence of the connectors, and (2) contact between adjacent surfaces forcing the buckling out-of-plane displacements to occur exclusively in a certain direction.

539 Out of all the possible buckling patterns compatible with the constraints imposed by the connectors and 540 contact between components, the one with the lowest buckling stress materialized. This resulted in some cases 541 in a buckling pattern with a varying half-wavelength along the member, as was observed in the outer channels 542 of columns SC2-6, or in a buckling pattern localized in one field, as seen in the plate sections of columns SC12 and SC1-3. The latter was only possible after some slip occurred between the components at the connectorpoints.

The experimentally derived buckling stresses were compared to theoretical predictions which considered the individual components in isolation, without any interaction with the rest of the cross-section, but used the experimentally observed wave-lengths. It was concluded that the buckling stress of the most slender components was increased by up to 44 % as a result of the restraint provided by the remainder of the crosssection. The amount of restraint was dependent on the connector spacing.

In terms of the cross-sectional capacity, columns SC1 exhibited a modest increase in ultimate strength of around 11 % when halving the connector spacing. This gain was mainly a result of the increase in the flexural buckling capacity of the plate sections in between connectors. When the critical buckling modes of the individual components were all local, the difference was even smaller. Only the SC2-2 columns showed around 11 % less capacity than the SC2-4 and SC2-6 columns, which had similar capacities. For columns SC3 and SC4 the effect of the connector spacing was negligible or non-existent.

#### 556 Acknowledgement

557 The authors are grateful for the financial support provided by the EPSRC through grant EP/M011976/1.

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Table 1: Nominal dimensions of the component sections

Column	section	h	b	l	t	$r_{int}$
Column	section	(mm)	(mm)	(mm)	(mm)	(mm)
SC1	T15414	154	54	-	1.4	2.8
SCI	P20024	200	-	-	2.4	-
5.02	T15414	154	54	-	1.4	2.8
<b>SC</b> 2	T7912	79	36	-	1.2	2.4
SC3/SC4	T12012	120	40	-	1.2	2.4
	S11012	110	50	10	1.2	2.4

Table 2: Measured dimensions of built-up columns 1

		Chann	el		Plate			
Column	section	h	b	t	section	h	t	
	section	(mm)	(mm)	(mm)	section	(mm)	(mm)	
SC1 20	T15414-1	153.96	53.90	1.452	P20024-1	199.83	2.476	
SC1-2a	T15414-2	154.19	53.60	1.440	P20024-2	199.73	2.466	
SC1 21	T15414-3	154.09	53.83	1.444	P20024-3	199.67	2.474	
SC1-20	T15414-4	154.09	53.43	1.435	P20024-4	199.83	2.472	
0.01.2	T15414-5	154.19	53.73	1.441	P20024-5	199.53	2.493	
SC1-5a	T15414-6	154.06	53.65	1.441	P20024-6	199.27	2.486	
SC1 21	T15414-7	154.13	53.41	1.429	P20024-7	199.43	2.482	
SC1-30	T15414-8	154.06	53.73	1.445	P20024-8	200.40	2.481	
SC1 5-	T15414-9	154.13	53.58	1.437	P20024-9	200.10	2.478	
SC1-5a	T15414-10	154.13	53.61	1.429	P20024-10	199.47	2.477	
0.01.51	T15414-11	154.09	53.35	1.417	P20024-11	198.93	2.472	
SC1-30	T15414-12	154.19	53.60	1.425	P20024-12	198.73	2.487	
Average		154.11	53.62	1.436		199.58	2.479	
St. Dev.		0.067	0.166	0.010		0.465	0.008	

 Table 3: Measured dimensions of built-up columns 2

		Chanr	nel		Channel			
Column	sastion	h	b	t	saation	h	b	t
	section	(mm)	(mm)	(mm)	section	(mm)	(mm)	(mm)
502.20	T15414-1	154.03	53.43	1.426	T7912-1	78.93	36.26	1.147
SC2-2a	T15414-2	154.06	53.28	1.416	T7912-2	78.99	36.25	1.172
SC2 21	T15414-3	153.96	53.46	1.428	T7912-3	78.83	36.40	1.142
SC2-20	T15414-4	153.96	53.40	1.407	T7912-4	79.13	36.30	1.176
SC2 4a	T15414-5	154.03	53.58	1.43	T7912-5	79.16	36.40	1.145
SC2-4a	T15414-6	153.96	53.26	1.438	T7912-6	79.09	36.46	1.169
SC2 4h	T15414-7	154.03	53.70	1.434	T7912-7	79.06	36.50	1.128
SC2-40	T15414-8	154.13	53.51	1.436	T7912-8	79.13	36.46	1.166
SC2 60	T15414-9	154.03	53.51	1.422	T7912-9	79.03	36.51	1.143
SC2-0a	T15414-10	154.29	53.35	1.417	T7912-10	79.06	36.41	1.172
SC2 6h	T15414-11	154.16	53.73	1.433	T7912-11	79.03	36.16	1.141
SC2-00	T15414-12	154.23	53.68	1.431	T7912-12	78.99	36.30	1.171
Average		154.07	53.49	1.427		79.04	36.37	1.156
St. Dev.		0.108	0.158	0.009		0.093	0.111	0.016

Table 4: Measured dimensions of built-up columns 3

		Channel				Channel			
Column	mn section $\begin{array}{c c} h & b & t \\ (mm) & (mm) & (mm) \end{array}$ section	saction	h	b	l	t			
Column		(mm)	(mm)	(mm)	section	(mm)	(mm)	(mm)	(mm)
502.2-	T12012-1 119.61 39.97 1.117 S11012-	S11012-1	110.46	49.83	9.83	1.109			
SC5-2a	T12012-2	119.97	40.01	1.090	S11012-2	111.07	49.93	9.87	1.095
6.02.01	T12012-3	119.82	40.07	1.102	S11012-3	110.75	49.79	9.83	1.107
SC3-20	T12012-4	119.84	39.99	1.097	S11012-4	110.91	49.92	9.88	1.090
SC2 5-	T12012-5	119.96	40.03	1.118	S11012-5	110.80	49.97	9.79	1.098
SC3-5a	T12012-6	119.81	40.01	1.127	S11012-6	110.44	49.93	9.89	1.119
CC2 51	T12012-7	119.59	39.99	1.124	S11012-7	110.07	49.90	9.87	1.120
SC3-30	T12012-8	119.72	39.95	1.095	S11012-8	110.85	49.82	9.85	1.098
Average		119.79	40.00	1.109		110.67	49.88	9.85	1.104
St. Dev.		0.144	0.035	0.014		0.325	0.064	0.033	0.011

Table 5: Measured dimensions of built-up columns 4

		Channel				Channel			
Column	section	h	b	t	section	h	b	l	t
Column	section	(mm)	(mm)	(mm)	section	(mm)	(mm)	(mm)	(mm)
SC4 2a	T12012-9	119.70	39.94	1.101	S11012-9	111.07	49.98	9.76	1.094
5C4-2a	T12012-10	119.90	39.98	1.089	S11012-10	111.13	49.91	9.86	1.088
SC4 2h	T12012-11	119.90	39.97	1.085	S11012-11	111.08	49.88	9.83	1.086
SC4-20	T12012-12	119.83	39.97	1.096	S11012-12	110.89	49.83	9.87	1.097
SC4 50	T12012-13	119.71	40.01	1.096	S11012-13	110.15	49.89	9.88	1.120
5C4-3a	T12012-14	119.89	39.98	1.096	S11012-14	111.11	49.86	9.88	1.103
CC4 5h	T12012-15	119.77	40.00	1.118	S11012-15	110.79	49.82	9.78	1.115
504-50	T12012-16	119.67	40.05	1.120	S11012-16	110.87	49.92	9.84	1.092
Average		119.80	39.99	1.100		110.89	49.89	9.84	1.099
St. Dev.		0.097	0.034	0.013		0.325	0.050	0.046	0.012

# Table 6: Material properties of tensile coupons

			1	Ξ	$\sigma_0$	.2%	0	u	8	f
Туре	Section	Coupon	(G	Pa)	(M	Pa)	(M	Pa)	(%	6)
			Ind.	Avg.	Ind.	Avg.	Ind.	Avg.	Ind.	Avg.
Flat	P20024	а	203	107	425	120	464	165	-	17
Flat	P20024	b	191	197	430	420	466	405	17	17
Flat	T15414	а	213	214	617	604	656	617	11	12
Flat	T15414	b	215	214	591	004	637	047	13	12
Flat	T7912	а	200	109	430	411	480	102	14	15
Flat	T7912	b	195	198	391	411	485	465	16	15
Flat	T12012	а	190	102	244	242	319	220	30	21
Flat	T12012	b	194	192	240	242	321	520	31	51
Flat	S11012	а	197	109	275	277	356	257	28	20
Flat	S11012	b	198	198	279	211	357	557	27	20
Corner	T15414	а	214	222	-	604	-	676	-	11
Corner	T15414	b	230	ZZZ	-	004	-	070	-	11
Corner	T7912	а	192	100	-	460	-	500	-	10
Corner	T7912	b	206	199	-	402	-	322	-	18
Corner	T12012	а	237	225		200		252		16
Corner	T12012	b	234	233		509		555		10
Corner	S11012	a	276	259		244		294		10
Corner	S11012	b	239	238		544		384		12

Specimen	Section	Imperfection (mm)				
			Max.	Avg.		
SC1	P20024	$\delta_{plate}$	0.60	0.21		
301	T15414	$\delta_{\scriptscriptstyle web}$	0.64	0.13		
	T7912	$\delta_{\scriptscriptstyle web}$	0.36	0.14		
SC2	T15414	$\delta_{\scriptscriptstyle web}$	0.69	0.15		
	113414	$\delta_{flange}$	0.47	0.19		
	T12012	$\delta_{\scriptscriptstyle web}$	1.04	0.20		
SC3		$\delta_{flange}$	0.58	0.30		
	S11012*	$\delta_{\scriptscriptstyle web}$	0.49	0.08		
		$\delta_{\scriptscriptstyle web}$	0.39	0.09		
SC4	S11012	$\delta_{flange}$	0.57	0.36		
5C4		$\delta_{\mathit{flange},L}$	0.06	0.01		
	T12012*	$\delta_{web}$	0.26	0.11		
*Imperfect	tions recorded	before the sec	tions were as	ssembled		

 Table 7: Maximum and average imperfection measurements

Table 8: Buckling stresses of the components of columns SC 1

Theoretic	cal buckli	Buckling stress			
	(MPa)	from test (MPa)			
C11	P	late	Channal	Plate	
Channel	Lower	Upper	Channel		
71	9	35	65	28	
71	9	36	72	19	
76	16	64	70	46	
75	16	64	69	45	
70	36	144	69	69	
69	36	145	67	67	
	Theoretic Channel 71 71 76 75 70 69	Theoretical buckli           (MPa)           P           Lower           71         9           71         9           76         16           75         16           70         36           69         36	$\begin{tabular}{ c c c } \hline Theoretical buckling stress \\ \hline (MPa) & (MPa) \\ \hline Channel & Plate \\ \hline Lower & Upper \\ \hline 71 & 9 & 35 \\ \hline 71 & 9 & 36 \\ \hline 71 & 9 & 36 \\ \hline 76 & 16 & 64 \\ \hline 75 & 16 & 64 \\ \hline 70 & 36 & 144 \\ \hline 69 & 36 & 145 \\ \hline \end{tabular}$	$\begin{tabular}{ c c c c } \hline Theoretical buckling stress & Buckling from test & from tes$	

Table 9: Buckling stresses of the components of columns SC2

	Theoretica	l buckling	Buckling stress from			
Column	stress	(MPa)	test (MPa)			
Column	Inner Outer		Inner	Outer		
	Channel	Channel	Channel	Channel		
SC2-2a	132	76	-	77-95		
SC2-2b	132	76	-	79		
SC2-4a	140	72	-	69-86		
SC2-4b	139	72	-	86		
SC2-6a	180	70-84	-	105		
SC2-6b	181	71-86	-	105		
Note:						
"-" indicates no estimation of the buckling stress was						
made as th	ne componer	nt was not the	e first one to	buckle.		

Table 10: Buckling stresses of the components of columns SC3

		Theoretica	al buckling	Buckling stress from		
Colu	Calumn	stress	(MPa)	test (MPa)		
Colu	11111	Plain	Lipped	Plain	Lipped	
		Channel	Channel	Channel	Channel	
SC3-	-2a	67-74	103	103	103	
SC3-	-2b	67-74	102	84-96	96	
SC3	-5a	70-77	104	-	96-117	
SC3	-5b	75	104	108	108	

Note: "-"indicates no estimation of the buckling stress was made as the component was not the first one to buckle

# Table 11: Buckling stresses of the components of columns SC4

	Theoretica	l buckling	Buckling stress from			
Column	stress	(MPa)	test (MPa)			
Column	Plain	Lipped	Plain	Lipped		
	Channel	Channel	Channel	Channel		
SC4-2a	66-73	100	87	-		
SC4-2b	66-72	100	88	-		
SC4-5a	66-73	105	73-91	91		
SC4-5b	69-76	103	93-104	104		
Note:						
"-" indicates no estimation of the buckling stress was						
made as th	ne componer	nt was not the	e first one to	buckle		

653
654
655

Table 12: Ultimate loads		
Column	Ultimate load (kN)	Average ultimate load (kN)
SC1-2a	183.97	176.07
SC1-2b	168.17	
SC1-3a	183.01	179.44
SC1-3b	175.86	
SC1-5a	201.72	195.11
SC1-5b	188.50	
SC2-2a	213.32	206.83
SC2-2b	200.34	
SC2-4a	238.00	235.70
SC2-4b	233.39	
SC2-6a	220.54	226.58
SC2-6b	232.62	
SC3-2a	139.30	138.92
SC3-2b	138.53	
SC3-5a	138.77	141.08
SC3-5b	143.40	
SC4-2a	148.09	147.56
SC4-2b	147.03	
SC4-5a	141.23	139.49
SC4-5b	137.74	









Figure 3: Location of connectors in: a) columns SC1; b) columns SC2; c) columns SC3 and d) columns SC4



Figure 4: a) M6 bolts, b) M5.5 self-drilling screws



Figure 5: Built-up columns after assembly: a) columns SC1, b) columns SC2, and c) columns SC3 and SC4



Figure 6: Corner coupons test set-up





Figure 7: Photograph of the cross-section of corner coupon T10412a







Figure 9: Imperfection measuring set-up







Figure 12: Potentiometer arrangement for: a) SC1; b) SC2; c) SC3 and d) SC4









Figure 14: Deformed shape approaching ultimate load in a) SC1-2a, b) SC1-3a, c) SC1-5a







Figure 16: Deformed shape approaching ultimate load in a) SC2-2a, b) SC2-4a, c) SC2-6a





Figure 19: Localised failure mechanism in column SC3-5b



Figure 20: Signature curve of the components of built-up columns 3 and 4



Figure 21: Deformed shape approaching ultimate load in a) SC4-2a, b) SC4-2b, c) SC4-5a, d) SC4-5b





Figure 22: Localised failure mechanism in column SC4-2b



Figure 24: Axial load vs. deformation curves for: a) columns SC1, b) columns SC2, c) columns SC3, and d) columns SC4