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1 Deformation of Armox 440T Plates Subject to Buried Explosive 2 Charge Detonations: A Benchmark for Appliqué Systems

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8 **Abstract**

9 Loading of vehicle undercarriages from the detonation of shallow-buried explosives remains a
10 serious threat to life in conflict and post-conflict zones. One method to protect lightly-armoured
11 vehicles is to retrofit them with appliqué armour, which must be strong enough to provide adequate
12 protection, but light enough to maintain vehicle manoeuvrability. A key performance metric of
13 this armour is its deformation under loading, which must be limited to avoid impact upon vehicle
14 occupants. The high-strength steel Armox 440T is commonly used due to its high load capacity,
15 strength-to-weight ratio, ductility and low cost: as other protection systems are developed, it would
16 be of great benefit to compare their deformation against an Armox 440T benchmark. However, no
17 definitive benchmarking study has been published to date, mainly due to the difficulties in ensuring
18 repeatable loading from complex buried detonations.

19 This paper presents experiments which underpin such a benchmarking study, building on the
20 authors' previous work to establish a methodology which produces very consistent loading from
21 shallow-buried detonations. Tests were conducted with a range of explosive masses and plate
22 thicknesses, with target plates secured in a purpose-designed frame to produce simple, consistent
23 boundary conditions. Plate deformations captured by stereo high-speed digital image correlation
24 were compared to a commonly-used low-cost peak deflection method. High-speed digital image
25 correlation was found to make highly reproducible displacement measurements with a standard
26 deviation of 2% of the mean. The low-cost method provided slightly higher variability up to 5%
27 of the mean value, and measurements of peak deformation were systematically 20% higher, but in

28 a consistent manner, with a low unit cost and without risk to expensive test equipment. The low-
29 cost method therefore allowed the development of a multivariate regression relationship between
30 deformation, charge size and plate thickness, which provides a benchmark for the assessment of
31 future protection solutions.

32 *Keywords:* Vehicle Protection, Buried charges, Landmines, Plate deformation, IEDs

33 **1. Introduction**

34 Blast loading of armoured materials is a subject of constant study, and the variation in types
35 of explosive charges and materials explored is necessarily vast, accounting for the uncertainty in
36 the type, size and location of potential threats to the armour. In the protection of armoured vehicle
37 undercarriages against ground-based threats, field trials are conducted to ascertain the potential
38 effects of the blast on the vehicle and, importantly, the personnel it carries. Armour affected by a
39 blast has a permanent, plastic residual deformation which is a retrospectively measurable quantity,
40 giving some indication of the damage to the vehicle and personnel. However, the peak elasto-
41 plastic displacement of the armour experienced during the blast is more relevant to the wellbeing
42 of the personnel within the vehicle, as this is the main cause of compression injuries. Therefore, a
43 reliable method for measuring and predicting the peak displacement of an armour structure from
44 buried explosive blast loading is of paramount importance.

45 With its combination of high yield-stress and elongation, Armox 440T has been widely studied
46 as a material for blast protection [1–6], but there is not yet a characterised relationship between
47 armour plate thickness and the displacement response to blasts. The setup described in this paper
48 lays out a modified experimental method which is relevant to military vehicles in theatre. From
49 the recorded deflection data an equation is presented for the maximal displacement response of
50 Armox 440T steel to buried explosive blasts from PE4 charges ranging between 400–1000 grams.

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51 **2. Background**

52 Military vehicles are an essential part of field operations, with over 4000 of these vehicles cur-
53 rently in use in the UK armed forces alone [7]. These vehicles are expensive and time-consuming
54 to design and deploy on military operations, and so adapting to ever-changing threats usually
55 means working with a vehicle already in service, adjusting it for improved performance in the
56 field. This can be achieved through the use of appliquéés. Hence, in order to tune the defence to
57 the threat, knowledge of the appliqué's performance is paramount.

58 One of the most dangerous and numerous threats to armoured military vehicles are buried
59 explosive devices, whether traditional mines or improvised explosive devices (IEDs). These im-
60 proved devices are a widespread threat, with at least 90 countries reporting recent attacks [8].
61 The main objective in armouring vehicles against blasts and impacts is to ensure the minimum
62 possible harm to personnel. In studies of active-duty personnel, up to 53% of injuries were due to
63 IEDs [9]. In particular, the blast loading and subsequent deformation of the floor of the vehicle ap-
64 plies significant axial loads to the passenger, which can cause permanent and debilitating injuries
65 to the lower leg [10], spine and neck [11]. The nature of these injuries emphasises the requirement
66 for armour material that absorbs as much of the energy of the blast as possible, and produces the
67 minimum peak deformation response to reduce compressional injuries.

68 Many studies have highlighted the effect of the geotechnical conditions surrounding a buried
69 charge on the impulsive output delivered to a target. Geotechnical parameters, such as moisture
70 content/saturation, air voids, bulk density, and particle size distribution, and physical parameters,
71 such as depth of burial and stand-off distance, are known to influence the impulsive output [12–
72 28]. Pickering et al. [22] showed that, for charges buried above a critical depth, impulse and the
73 resulting deformation of a steel plate generally increase with increasing depth of burial. Once this
74 critical depth is reached, the impulse continues to increase with the burial depth, but the plastic
75 deformation of the plate then starts to linearly decrease with increased depth. Clarke et al. [24]
76 presented the factors that affect the repeatability of buried charge testing, and highlighted the
77 importance of keeping the test soil as uniform as possible with the water content tightly controlled,
78 contrary to the belief that density alone is enough to predict the output of an experiment [14].

79 For structural applications there has also been much interest in quantifying the deformation
80 of steel plates under the impulse imparted by bare charges [29–32], with recent improvements in
81 technology allowing the elastic, transient response to also be captured [3, 33, 34]. Testing with
82 buried charges is not often included in these test programmes due to the increased complexities
83 involved, although there are notable exceptions [25, 26]. The current work adapts the methodology
84 used previously [25], eliminating the direct measurement of impulse in favour of creating the most
85 repeatable test conditions for armour systems. For the application of armoured material as an
86 appliqué to existing military vehicles, a parametric study of the peak deformation of armoured
87 steel from buried charges is essential.

88 This paper builds on previous research, using both mechanical and optical methods to mea-
89 sure the peak deformation of various thicknesses of ArmoX 440T steel plate exposed to a range
90 of charge sizes. An experimental methodology is also presented which is known to deliver a re-
91 peatable distribution of impulse from a buried charge, thus defining a standardised experimental
92 methodology for measuring the relative performance of future appliqué systems.

93 **3. Methodology**

94 The results of two separate test series are reported, one parametric study which measured plate
95 deformations using aluminium crush block and one focussed study which (unsuccessfully) sought
96 to validate heterodyned velocimetry measurements¹ against digital image correlation. Because the
97 experimental methodology was held constant between studies, the results are comparable and the
98 digital image correlation results can be used to provide time resolved deformation data under the
99 same test conditions.

100 The experimental work was conducted by Blastech Ltd. at the University of Sheffield Blast &
101 Impact Laboratory, Buxton, UK as part of a research project funded by the UK Defence Science
102 and Technology Laboratory (Dstl). Armour targets, approximately half-scale models of armoured
103 vehicle dimensions, were mounted to a custom-designed rigid reaction frame to provide a con-

¹Heterodyned velocimetry is a non-contact method of measuring the velocity-history of a moving object. It has been used with to measure explosively driven plates [35] and was being evaluated as a possible technique for measuring the deformation of complex structures where other measurement techniques were not possible.

104 sistent boundary condition for plate deformation measurements. Half-scale models were used
105 to allow a future comparison of alternative material types, such as polymer composites [36] or
106 sandwich structures [37] and obviate some of the observed difficulties in small-scale testing of
107 such structures [36]. The frame was constructed from large-section rectangular tubing, braced to
108 increase stiffness as shown in Figure 1. The rigid test frame was modified from that previously de-
109 scribed in [24] by welding additional stiffening to the underside of the frame, to allow attachment
110 of the target interface plate.

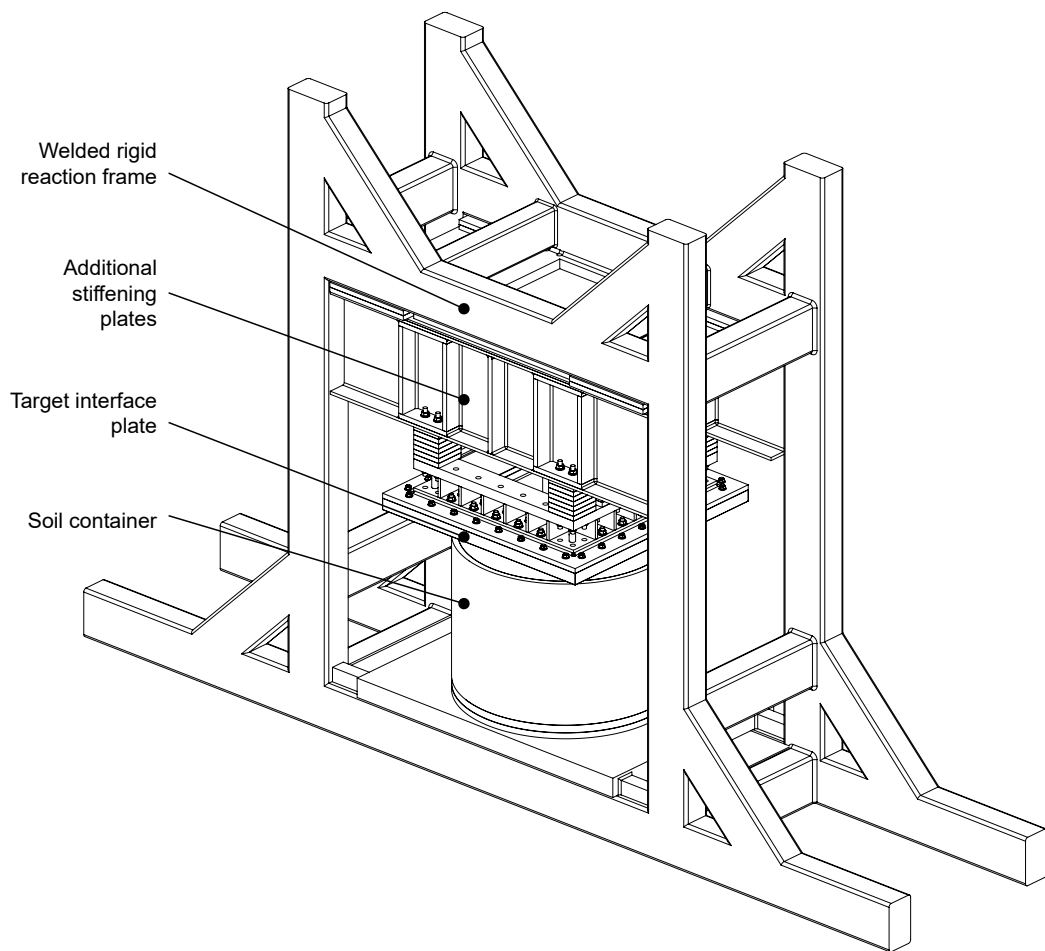


Figure 1: Overview of test apparatus

111 The target interface plate consisted of a fully-welded square ‘picture-frame’ consisting of a
112 1205 mm square flat plate, with the centre cut out to form a 655 mm square open aperture (Fig-
113 ure 2). Four trapezoidal 65 mm thick S355 steel plates were bolted to the underside of the picture

114 frame with M20 shoulder bolts and cap-head bolts to form a flat surface onto which to mount
 115 targets. The inside edges of the trapezoidal plates were machined to a 25 mm radius to reduce the
 116 shear-stress in the target plates [38].

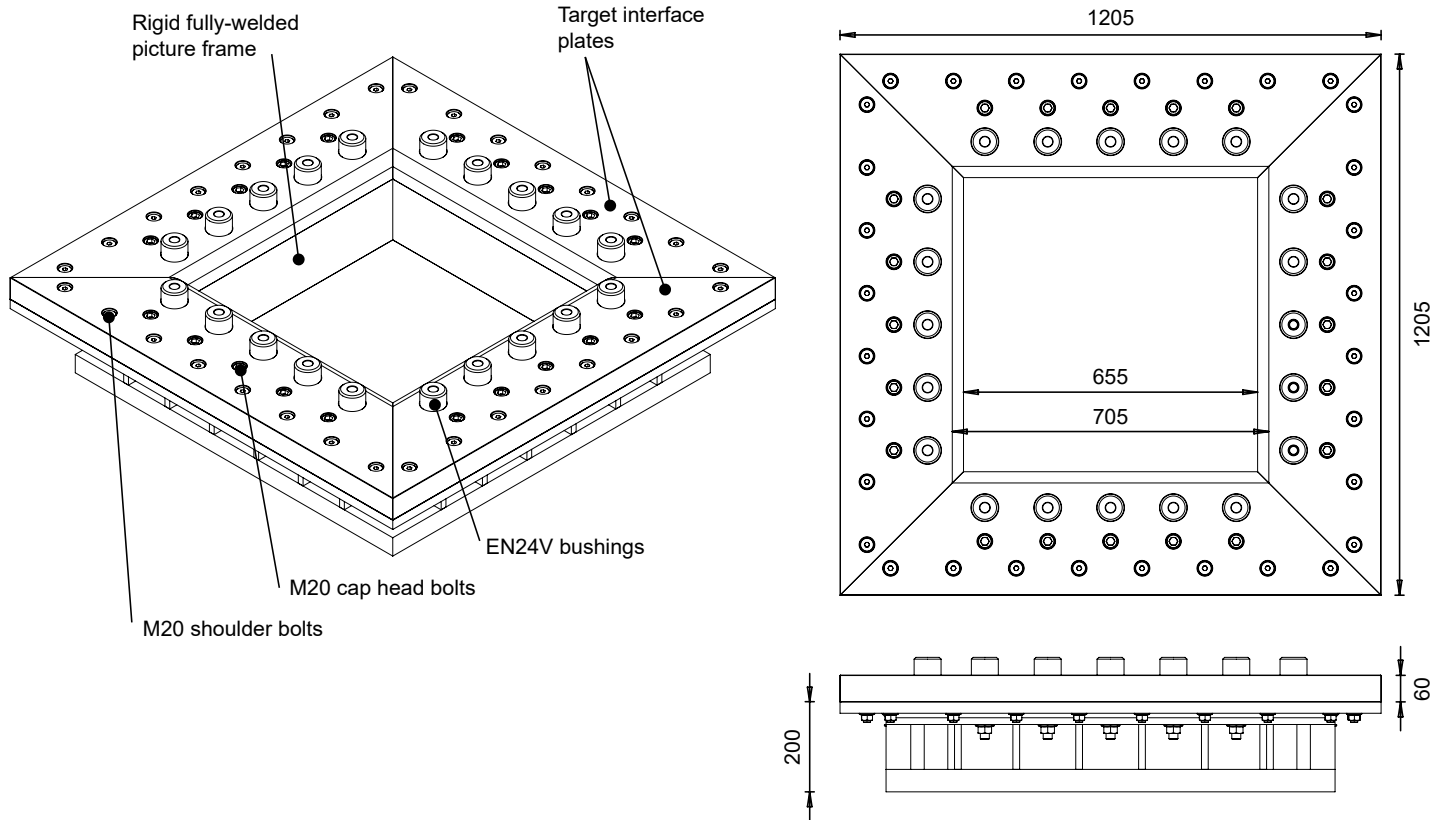


Figure 2: Schematic of the target interface plate (viewed from the underside)

117 Along the inside edge of each of the trapezoidal target interface plates, five 60 mm diameter
 118 EN24V bushings (twenty in total) protruded 40 mm from the underside of the flat mounting sur-
 119 face. These shoulder bushings fit into machined pockets in the trapezoidal plates and were there-
 120 fore held captive against the lower surface of the fully welded picture frame. A 20 mm clearance
 121 hole was drilled through the centre of each bushing allowing an M20 bolt to be inserted through
 122 the hole into a captive nut on the upper surface of the picture frame as shown in Figure 3. The
 123 design of the interface plate was the culmination of an iterative design process over many years.
 124 Experiences from earlier test campaigns were that slip boundary conditions resulted in variability
 125 in deformation that was not particularly well correlated with the consistency in total impulse [25]

126 and this test apparatus was designed on the postulate that better control of the boundary condition
127 would lead to more reproducible results.

128 Target plates consisted of 995 mm square plates of ArmoX 440T steel in various thicknesses
129 representing half-scale appliqué armour; the thickness of each plate was measured with callipers
130 prior to each test. The target plates were prepared by cutting 64 mm diameter holes into the plate
131 to fit over the 60 mm bushings. Target plates were lifted to fit flush against the trapezoidal plates
132 and spacers positioned around each of the bushings, of such a thickness to protrude below the
133 lower extent of the bushing (Figure 3). A flat washer was positioned below the spacers and M20
134 bolts were passed through the bushings and tightened against the captive nuts. Depending on the
135 thickness of the targets, more or fewer spacers were used to ensure the targets were held against
136 the interface plates. By adopting this approach, the bolts holding the target to the test frame were
137 solely in tension, while the bushings supported the shear loads from the target pulling inwards
138 under blast loading; because of the large diameter of the bushings, the shear stress was sufficiently
139 low to prevent plastic deformation. In this way a consistent boundary condition for the targets was
140 ensured and the variation in target response between tests was minimised.

141 A 90 mm × 90 mm, 200 mm tall 5.2-1/4-25N-3003 aluminium honeycomb crush block was
142 placed directly behind the target and braced against a rigid reaction beam, which was bolted to the
143 test frame above the crush block. This system was used to measure the peak dynamic deflection of
144 rear face of the target. The strength of the material is quoted as 4.3–4.6 MPa (therefore initiating
145 crushing at a force of 35–37 kN) with a prolonged crush stress of 1.6 MPa (applying a consistent
146 force of 13 kN) [39, 40]. To improve the positioning of the crush block, the sides of the block
147 were wrapped in adhesive tape to provide a better adhesion when the crush block was taped to the
148 underside of the rigid reaction beam.

149 *3.1. Geotechnical conditions*

150 Controlling soil parameters has been shown to effectively reduce the variability of buried
151 charge experiments and so the methods of filling the container with soil were replicated from
152 previous studies [24]. Soil containers were constructed from 30 mm thick rolled mild steel plate
153 formed into a 1000 mm internal diameter, 750 mm tall cylinder, with a 50 mm thick mild steel

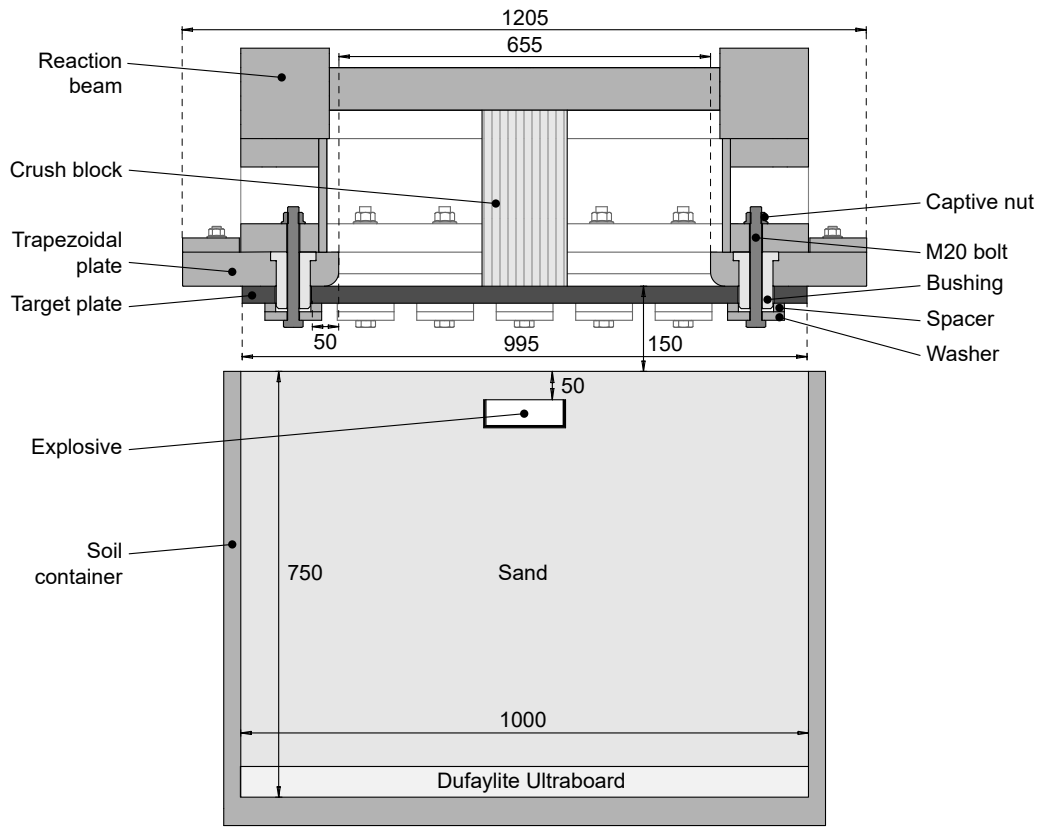


Figure 3: Sectional view through the test apparatus

154 plate welded to the base. A layer of cardboard energy-dissipating material was added into the base
 155 of the soil bin; three layers of 18 mm thick Dufaylite Ultraboard were cut to size and placed in
 156 the base of the soil bin and then covered with plastic sheet to protect the cardboard from moisture.
 157 The reason for this addition was to protect the base of the soil bin from plastic deformation.

158 The soil containers were filled with Leighton Buzzard 14/25 sand, as was used in previous
 159 studies [24]. As well as producing good levels of repeatability, explosive charges buried in this
 160 sand have been studied extensively to measure the reflected pressure imparted to flat targets [25,
 161 41] which provides useful information for follow-on simulations.

162 The containers were filled in three lifts. The initial moisture content of each lift was checked
 163 prior to filling, and the mass of water required to bring the moisture content to $5 \pm 2\%$ by mass was
 164 added to the sand. The sand was mixed in a forced action pan mixer until the water was evenly
 165 distributed. The moisture content was then re-checked and, if the moisture content was confirmed

166 to be within tolerance, the contents of the pan mixer were purged into the soil container, taking
167 care to avoid sample loss. Plywood shuttering (cut to the internal diameter of the container) was
168 placed on the surface of the soil and a stiffened steel plate (100 mm clear of the internal diameter)
169 was seated on the timber boards. A vibrating compaction plate was placed upon the stiffened steel
170 plate and the soil vibrated until it reached its target level of compaction based on the soil depth
171 measured in the container. The vibrating compaction plate, stiffened steel plate and timber boards
172 were then removed from the container with care such that the soil surface remained undisturbed.
173 This was repeated until the container was filled. The density was then calculated based upon the
174 internal dimensions of the soil container and mass of soil and water added. The target density for
175 the sand was 1.62 Mg/m^3 for all tests and was achieved to within 1.45%.

176 3.2. Explosive setup

177 PE4 charges were buried in the sand. The charges were each 3:1 cylinders, packed into open-
178 topped, 3D printed, PLA plastic cases with 4 mm thick walls and bases. The charges ranged in size
179 from 400 g to 1000 g. During commissioning tests, the charge size was increased to induce defor-
180 mation of between 50 mm and 100 mm in the targets. The lower limit was to allow measurement
181 errors arising from surface imperfections in the deformed crush block to be small in comparison
182 to the magnitude of the deformation measured. The upper limit was selected after observing a per-
183 foration of a target where similar plate thicknesses and charge masses were observed to undergo
184 approximately 100 mm of deformation. Each charge was buried with an overburden of 50 mm
185 of sand above the charge. To place the charge, a cavity was excavated which was approximately
186 5 mm wider than the charge geometry (depth was set precisely). Excavated material was placed
187 into a sealed bag so that it remained at the correct moisture content. The soil containers were
188 placed under the targets such that the stand-off distance was 150 mm for all tests, as measured
189 from the soil surface to the rear of the target plate. By holding the distance to this plane constant,
190 any improvements in the structural response of thicker targets must be suitably large to outweigh
191 the increased loading experienced as a result of being closer to the detonating explosive, as would
192 be the case for appliqué systems in operation. A non-electric detonator was inserted into the top
193 of the explosive charge, and the soil was replaced above the charge. The mass of soil remaining

194 in the bag was then weighed to determine that the density of the replaced soil was the same as the
195 excavated material. A sectional view through the test arrangement is shown in Figure 3.

196 3.3. Deflection measurement

197 Residual deflections were measured by generating 3D models of the target plates using a pho-
198 togrammetric technique. For each plate 16 photographs were taken and processed into a point
199 cloud using VisualSFM. This point cloud was then scaled to real world units in MeshLab using
200 the length of the plate edge. Finally, MATLAB was used to mesh the point cloud to a 10 mm grid
201 from which the position of points can be inferred. In the source photographs the average pixel
202 represents a length of approximately 2.5 mm, and the plate edge can be identified in the point
203 cloud within approximately 5 mm; on a 995 mm plate this corresponds to a maximum scaling
204 error of $\pm 1.5\%$ on the deflection measurements. An example of the residual deflections calculated
205 from this analysis is shown in Figure 4 for a 12 mm thick plate. The peak centre-point deflections
206 (before the plate relaxed to the profile of the type seen in Figure 4) were measured as the distance
207 from the deformed aluminium crush block surface to the original location of the rear target face.

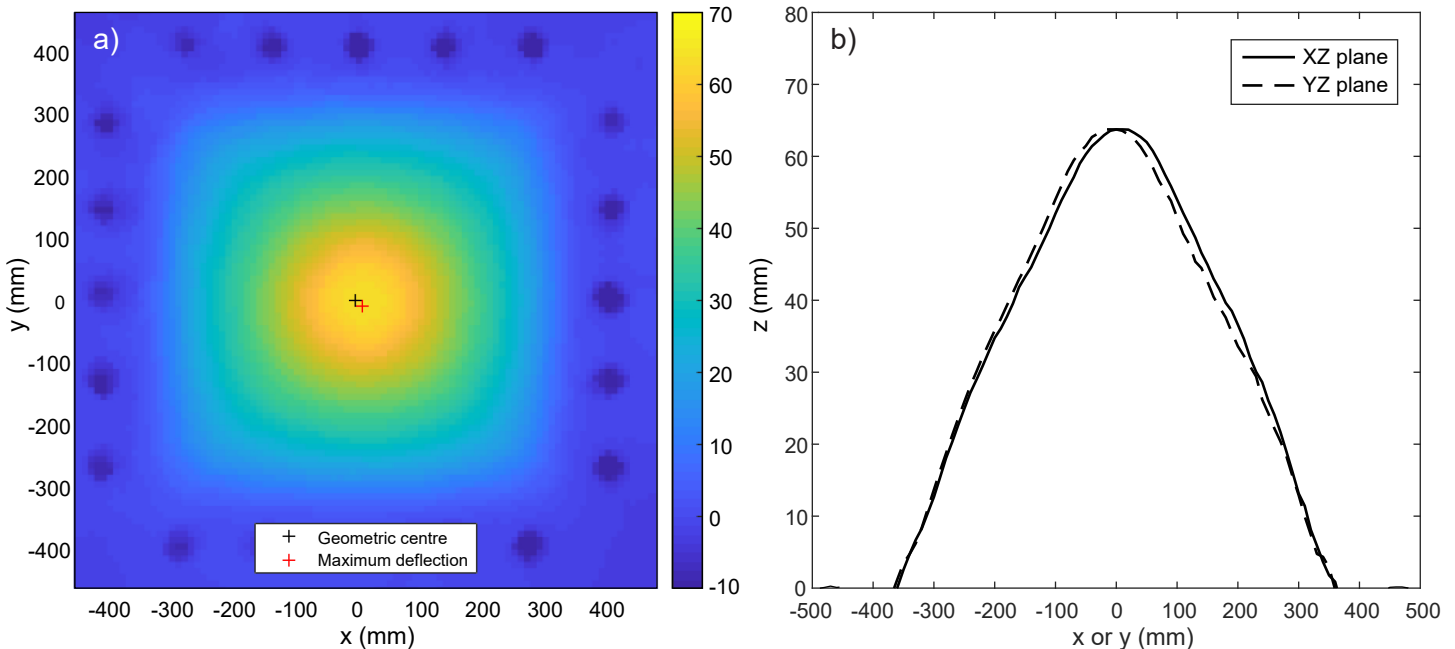


Figure 4: Example test results for a 12 mm plate against a 1000 g charge showing a) a contour plot of the deformed plate profile and b) cross-sections through the centre of the deformed profile, as measured after the test

208 4. Validation

209 One of the main advantages of using aluminium crush block for deflection measurements is
210 its ease of deployment in aggressive environments. To gauge the absolute accuracy of the crush
211 block system, a validation exercise was undertaken using high-speed digital image correlation
212 (DIC) of the back face of the target plate. Due to the risk posed to the optical system from
213 plate perforation, testing was conducted with arrangements which provided a large peak deflection
214 but were known not to perforate the target plate. It was observed during testing that conducting
215 DIC measurements in an outdoor environment was challenging, particularly in contrast to studies
216 conducted indoors with much smaller test samples [42]. Video cameras had to be removed from
217 the test apparatus at the end of each day to prevent environmental damage and the subsequent
218 installation and recalibration reduced the number of tests which could be undertaken in any given
219 day. Testing was impossible in wet weather conditions, owing to moisture settling on the target
220 plate and spalling off the surface after detonation, obscuring the cameras' view. While select data
221 points were obtained with DIC the practical considerations of testing in this way made it unfeasible
222 to replicate the number of data points obtained with crush blocks.

223 Three tests were conducted with 12 mm thick Armox 440T targets against 1000 g PE4 charges.
224 To facilitate this, the rigid reaction beam holding the crush-block was removed and a steel I-beam
225 was inserted through the reaction frame's structure in its place. The I-beam was independently
226 supported on concrete blocks so that it was not in contact with the reaction frame and therefore
227 would experience lower transient accelerations during the tests. Two Photron SA-Z high speed
228 cameras were bolted directly to the I-beam and oriented with a point 50 mm from the centre of
229 the target plate. The angle between cameras was approximately 50° as shown schematically in
230 Figure 5. LED lights were mounted adjacent to the cameras to provide illumination during the
231 test.

232 Prior to each test, the upper face of the targets was painted with a high-contrast random pattern
233 to facilitate DIC from the two cameras [43]. Targets were sand-blasted to remove surface corrosion
234 and mill-scale and wiped clean. A thin layer of white paint was then sprayed onto the upper surface
235 and allowed to dry. Black paint was then sprayed onto a piece of stainless steel wool and lightly

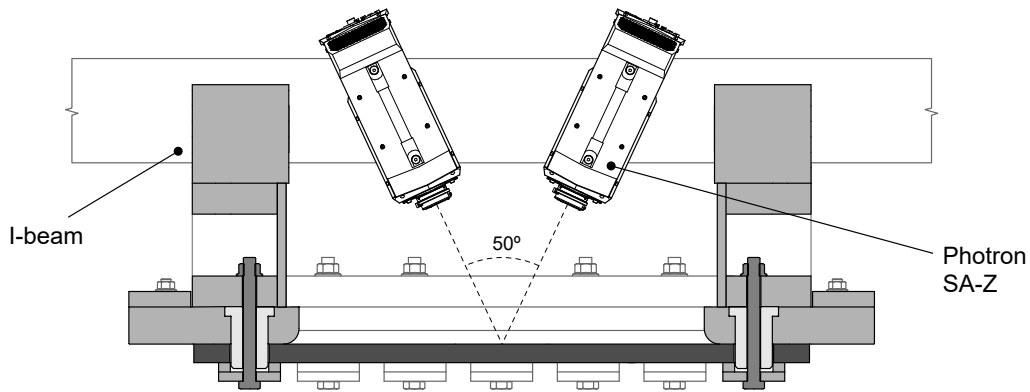


Figure 5: Experimental set-up for temporally-resolved DIC measurements

236 applied to the white surface to transfer a randomly-shaped patch of black paint. This process was
 237 repeated until the upper surface was approximately half covered with patches of black paint. A
 238 30 mm diameter area at the centre of the plate was not covered with paint, and so the DIC could
 239 make no measurements in this region. (This area was used as a reflective surface during a trial of
 240 heterodyned laser velocimetry as an alternative method of displacement measurement.)

241 After the target plate was installed, the cameras were calibrated by placing a calibration board
 242 in the cameras' fields of view. A test image of the target plate was then recorded to confirm that
 243 the calibration had been successful and that the paint pattern's coverage was sufficient to allow the
 244 displacement field across the plate's surface to be mapped; additional black shapes were added in
 245 regions of the plate where the software could not resolve the plate's shape.

246 The cameras were then set to synchronised recording of the plate at 20,000 frames per second
 247 and $1,024 \times 1,024$ pixel resolution across the central 400 mm of the plate. They were triggered
 248 to start recording by a fibre-optic light sensor embedded in the PE4 charge, which output a TTL
 249 signal upon detonation. Typically the cameras recorded for 5 ms before entrained sand encroached
 250 around the target and obscured the cameras' views of the plates. The video footage was then
 251 processed using ARAMIS to establish the upward deformation at a range of points across the
 252 upper surface of the plate. The deformation of a point 50 mm away from the target centre (therefore
 253 within 5 mm of the footprint of the crush block position) is plotted from three tests in Figure 6,
 254 and it can be seen that the results are consistent between tests throughout the deformation. The
 255 late time DIC data also provides an additional validation for the residual deflection recorded using

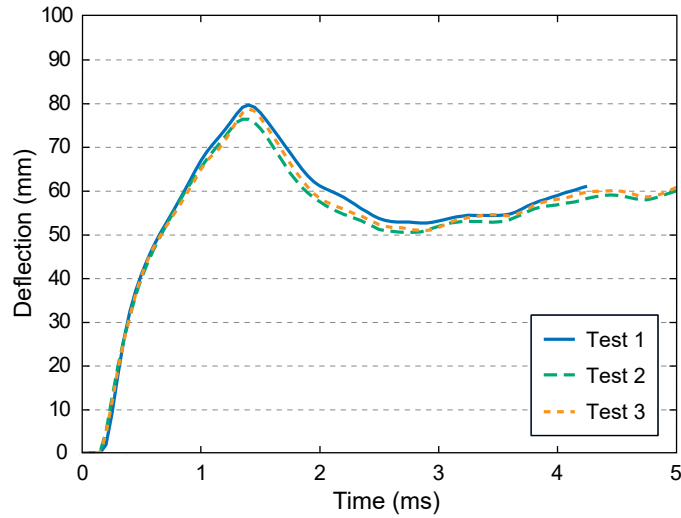


Figure 6: Upward displacement of a point 50 mm from the centre of a 12 mm thick ArmoX 440T plate, loaded by the detonation of a 1000 g buried PE4 charge

256 photogrammetry (Figure 4). From Figure 4b, the residual deflection at an offset of 50 mm from
 257 the plate centre is ≈ 61 mm, which agrees well with the late time deflection data in Figure 6.

Table 1: Peak displacement validation data

Method	Plate thickness (mm)		Areal density (kg/m ²)	Charge mass (g)	Peak centre-point deformation (mm) †		
	Nominal	Measured			δ	\bar{x}	s
DIC	12	12.4*	96.9	1000	79.5	78	1.7
		12.4*			76.3		
		12.4*			78.6		
Crush block	12	12.3	96.6	1000	95	94	1.7
		12.3			95		
		12.4			92		

Notes: *Average of all 12 mm plates tested was used (12.35 mm),
 † 50 mm offset for DIC, \bar{x} = mean, s = standard deviation

258 The peak deformation measured in each test is summarised in Table 1, and is compared against
 259 the equivalent crush block data. The precise thickness of the plates for the DIC tests was not ex-
 260 plicitly measured, and so the plate thickness is inferred from the mean of all the 12 mm plates used
 261 in the crush blocks tests (see Table 2). The average peak deformation measured using DIC was

262 78 mm, with a standard deviation of 1.7 mm (2% of the mean). This standard deviation compares
263 well with the three crush block tests with 12 mm plates and 1000 g charges, but peak deformation
264 measurements made using the crush blocks were systematically higher than comparable tests using
265 DIC, by an average of 16 mm. A heteroscedastic two-tailed Student's T-test returned a very low
266 probability (0.03%) that the two test methods have the same sample mean (although with small
267 numbers of test points this finding should be treated with caution). One reason for the discrepancy
268 between the DIC and crush block data could be the difference in measurement locations, with
269 the DIC measuring at an offset of 50 mm from the plate centre. Assuming the peak and residual
270 deformed shapes are similar, it is possible to calculate that measurements at an offset of 50 mm
271 would lead to displacement readings approximately 5 mm lower than at the true plate centre. It is
272 also hypothesised that the initial movement of the target plate imparted momentum to the struc-
273 ture of the crush block, causing it to continue to compress after the plate itself had reached its
274 maximum deformation and begun to return to its final position (as seen in Figure 6). This would
275 mean that the crush block underwent higher levels of deformation than had actually occurred in
276 the plate, leading to a conservative measure of peak deflection.

277 To investigate the credibility of this hypothesis simplified simulations were conducted using
278 LS-Dyna. The target geometry was replicated from Figure 3 using the existing material models
279 for the plate material [2]. A 44 mm × 32 mm, 150 mm tall crush block was added at an initial
280 1 mm standoff from the plate and constrained at the upper surface; while not perfectly match-
281 ing the experimental geometry, this was deemed sufficiently close to investigate the phenomenon
282 without building and meshing new components. The plate was caused to deform using the *INI-
283 TIAL_IMPULSE_MINE card with a 500 g TNT 3:1 aspect ratio mine buried 72 mm deep in a
284 1674 kg/m³ sand 150 mm below the target. The input parameters were manually varied until the
285 initial velocity of the plate matched the DIC experiments to within 20%. The maximum plate
286 deformation in the simulated case was 25% lower than the experimental case. The crush block
287 was modelled using shell elements following an approach described by Jost et al. [44]. The hon-
288 eycomb was constructed from 0.04 mm thick 5052 aluminium foil with a cell-size of 5 mm, using
289 material models from Panicker et al. [45]. The deformation predicted in the target and lower sur-
290 face of the crush-block are compared to the average experimental deformations in Figure 7. It can

291 be seen that, although the simulated plate deformations do not match those observed experimen-
292 tally, the crush block is still predicted to continue crushing after losing contact with the target; the
293 crush-block final deformation exceeds the maximum deformation of the target by 3 mm. Although
294 the simulations are simplifications of the experimental setup, they show that proposed hypothesis
295 of over-crushing in the crush-block is credible and go some way to explaining the discrepancy
296 between crush-block and DIC results.

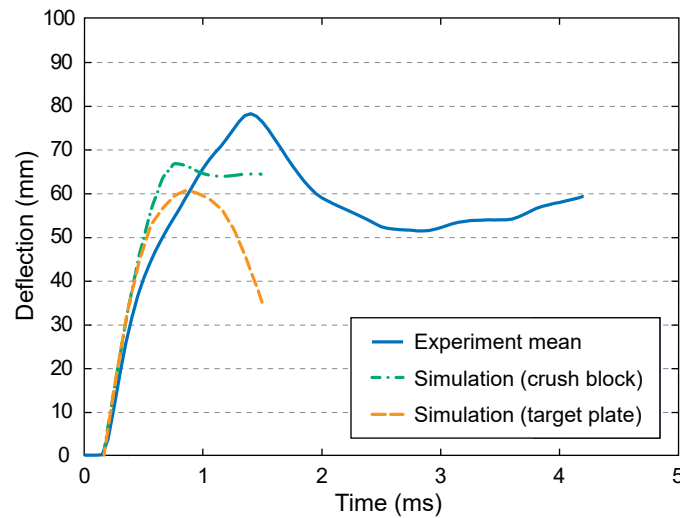


Figure 7: Average displacement of a 12 mm thick ArmoX 440T plate and comparative simulations of the deformation of the same plate and the consequent displacement of the lower surface of the crush block

297 Despite this discrepancy, the crush block data has been shown to be a repeatable measure
298 of peak deformation, for the purposes of providing a systematic comparison between appliqué
299 systems. Crush blocks have the additional benefits over DIC of being low-cost, quick to deploy
300 and being easy to replace in case of target perforation.

301 5. Results

302 With the validation exercise complete, twenty four tests were conducted with crush-block peak
303 deflection measurements using a range of charge masses and plate thicknesses, which are sum-
304 marised in Table 2. The charge size was initially chosen to be 400 g and then increased until
305 appreciable deformation of the targets was observed, but without causing target rupture or dam-
306 age to the test fixture. (One target was observed to rupture: a 6.2 mm thick plate subject to the

307 detonation of an 800 g charge.) The areal density of each target was calculated by multiplying the
 308 measured thickness by the density of the steel which was taken as 7850 kg/m³ [2]. Areal density
 309 was used in preference to plate thickness to allow a future assessment of materials of different den-
 310 sities. This approach should allow the deformation induced in, for example, an aluminium alloy
 311 plate to be measured and an assessment made of whether replacing Armox 440T armour plates in
 312 armour designs could allow deformation to be reduced while maintaining the same armour mass.

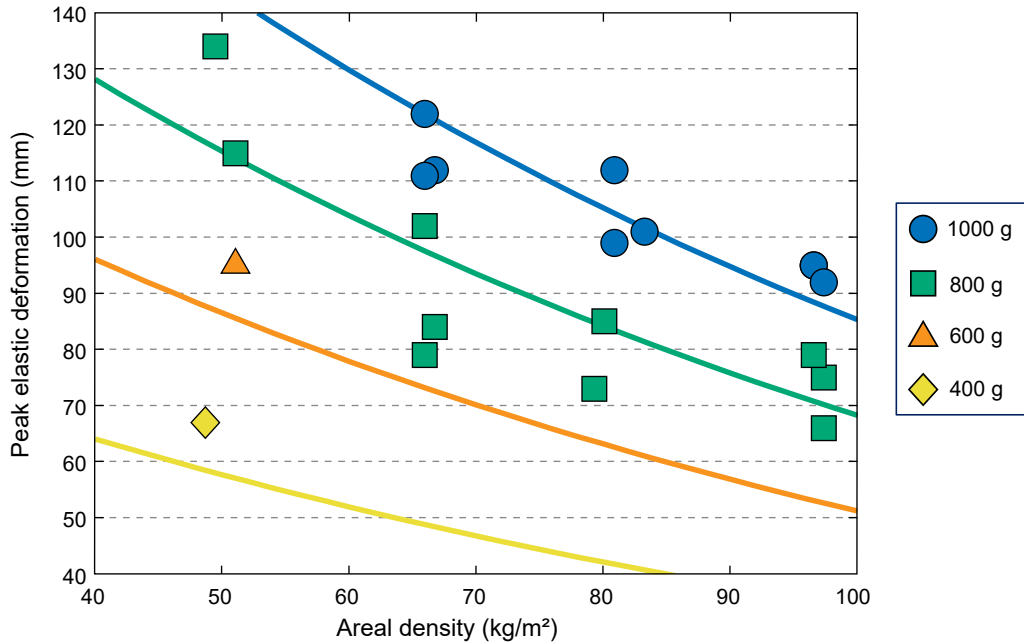


Figure 8: Peak deformation (measured from crush block compression) of Armox 440T as a function of target areal density for different explosive masses. Contours of the curve fits to the data (Equation 1) for charge masses of 400 g, 600 g, 800 g, and 1000 g have also been superimposed

313 The peak deformation values from Table 2 are plotted in Figure 8, where it can be seen that
 314 the peak deformation increased with explosive charge mass and decreased with areal density. It
 315 was observed that deformation levels in Figure 8 increased linearly with explosive mass (which is
 316 supported by [26]) and decreased approximately exponentially with areal density. To interpolate
 317 between data points, a single curve of the form in Equation 1 was fitted to all of the data:

$$D = aMe^{-bA} \quad (1)$$

318 where D is the peak deformation, M is the explosive mass and A is the areal density of the target

Table 2: Summary of peak and residual deformation results.

Plate thickness (mm)		Areal Density (kg/m ²)	Charge mass (g)	Peak centre-point deformation (mm)			Residual centre-point deformation (mm)		
Nominal	Measured			δ	\bar{x}	s	δ	\bar{x}	s
6	6.2	48.7	400	67	-	-	60	-	-
6	6.5	51.0	600	95	-	-	78	-	-
6	6.5	51.0	800	115	125	13.4	98	99	2.9
	6.2	48.7		*			97		
	6.3	49.5		134			102		
	6.2	48.7		†			†		
8	8.4	65.9	800	79	88	12.1	71	73	1.3
	8.5	66.7		84			73		
	8.4	65.9		102			74		
10	10.1	79.3	800	73	81	6.9	57	57	0.5
	10.2	80.1		85			57		
	10.2	80.1		85			58		
12	12.3	96.6	800	79	73	6.7	49	49	0.6
	12.4	97.3		66			50		
	12.4	97.3		75			49		
8	8.5	66.7	1000	112	115	6.1	93	91	4.4
	8.4	65.9		111			86		
	8.4	65.9		122			94		
10	10.6	83.2	1000	101	104	7.0	74	73	1.2
	10.3	80.9		112			72		
	10.3	80.9		99			72		
12	12.3	96.6	1000	95	94	1.7	62	63	0.9
	12.3	96.6		95			64		
	12.4	97.3		92			62		

Notes: *No data recovered, † Target rupture, \bar{x} = mean, s = standard deviation

319 armour. The form of the function in Equation 1 was chosen such that the deformation tended to
 320 zero as explosive mass tended to zero and areal density tended to infinity; this was assessed as
 321 more likely to permit small amounts of extrapolation without yielding non-physical predictions. A
 322 multivariate least squares method was used to fit Equation 1 to the data and determine the constants
 323 a and b . An error function (E) was calculated from the 24 data points (labelled i) using Equation 2.

$$E = \sum_{i=1}^{24} (aM_i e^{-bA_i} - D_i)^2 \quad (2)$$

324 A Generalized Reduced Gradient non-linear solver was then used to arrive at a numerical approxi-
 325 mation of the constants a (0.24367 mm/g) and b (0.01050 m²/kg) which minimised the size of the
 326 error (E). Using these values, contours of $M = 400$ g, 600 g, 800 g and 1000 g are superimposed
 327 on the data in Figure 8.

328 As there are more data points at 800 g and 1000 g charge-weights, the curve-fitting algorithm
 329 preferentially fits the data at those charge masses. Consequently, the curves described by Equa-
 330 tion 1 are a better fit to the higher charge weights than the 400 g and 600 g data points. Because
 331 Equation 1 is not linear (and cannot be readily linearised), many of the conventional methods for
 332 analysing the quality of a data fit are not appropriate and confidence intervals cannot be applied
 333 to Figure 8. An estimate of the confidence intervals, however, can be made by treating the 800 g
 334 and 1000 g data sets as independent data with constant M , linearising Equation 1 and using linear
 335 regression to apply confidence intervals to the data. This is shown in Figure 9. These confidence
 336 intervals are likely to be similar to the confidence which can be applied to the data fit shown in
 337 Figure 8.

338 The quality of the fit could likely be improved by modifying the form of Equation 1 but, in the
 339 absence of a physical justification for introducing additional degrees of freedom into the model, it
 340 was deemed to be sufficiently accurate to form a baseline appliqué benchmark.

341 The benefit of fitting a single curve to the data is that performance of a different armour can
 342 be compared to the Armox 440T baseline. If a test is performed on a new armour with areal
 343 density A , and the detonation of a charge mass, M , induces a deformation D , Equation 1 can be
 344 solved to determine the areal density of Armox 440T which would experience the same level of

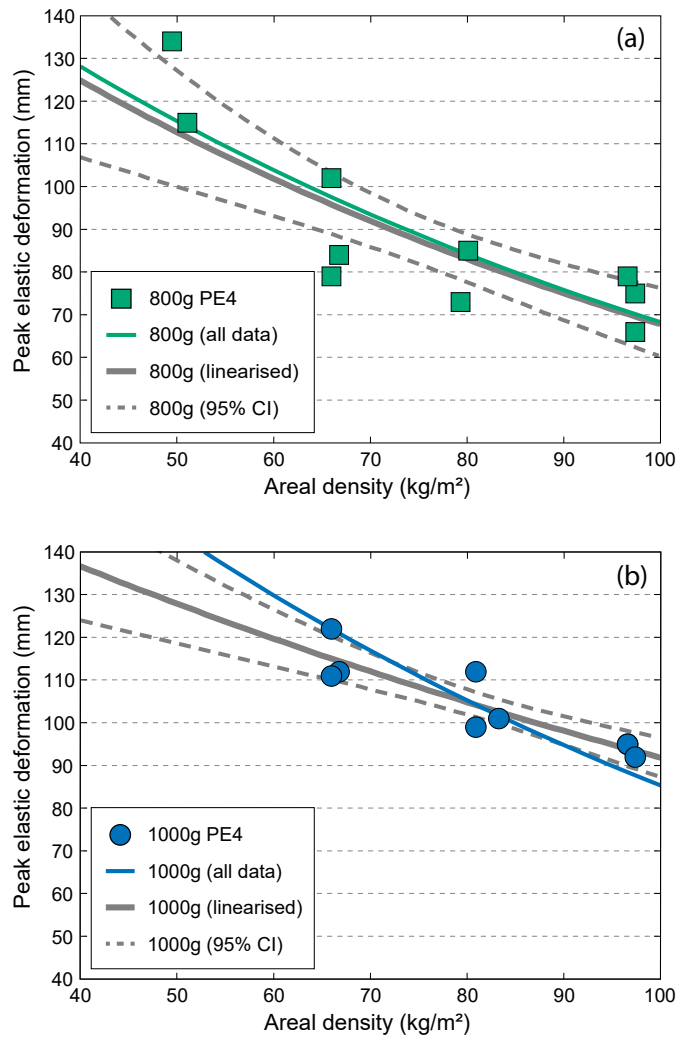


Figure 9: Peak deformation (measured from crush block compression) of ArmoX 440T as a function of target areal density for a) 800 g and b) 1000 g explosive masses. Regression lines from the full data fit (all data) and linearised fit with constant M (linearised) along with associated 95% confidence intervals (95% CI) have also been superimposed

345 deformation. In this way the level of mass saving which could be realised by the new armour can
 346 be estimated. Alternatively, Equation 1 can be solved for M to estimate to what extent the new
 347 armour will increase the protection level offered.

348 The residual plastic deformation showed much lower levels of variability than the peak elasto-
 349 plastic deformation (Table 2). The standard deviation on the residual deformations was 1-4 mm
 350 (1-5% of the mean value for equivalent tests), compared to the standard deviation on the peak
 351 deformations of 1-13 mm (2-14% of the mean). This implies that the measurement of residual

352 deformation is more reliable than the peak deformation measured using the aluminium crush block.

353 The residual plastic deformation is strongly correlated to peak elasto-plastic deformation (Fig-
354 ure 10), but does not appear highly correlated with other factors. Fitting a least-squares trend line
355 to all the data shows a gradient of 0.77 (and a constant offset of 2 mm). As a result, the proportion
356 by which a plate relaxes to its final, plastically deformed, state is relatively constant over the levels
357 of deformation observed; plates appear to relax by between 15 mm and 35 mm.

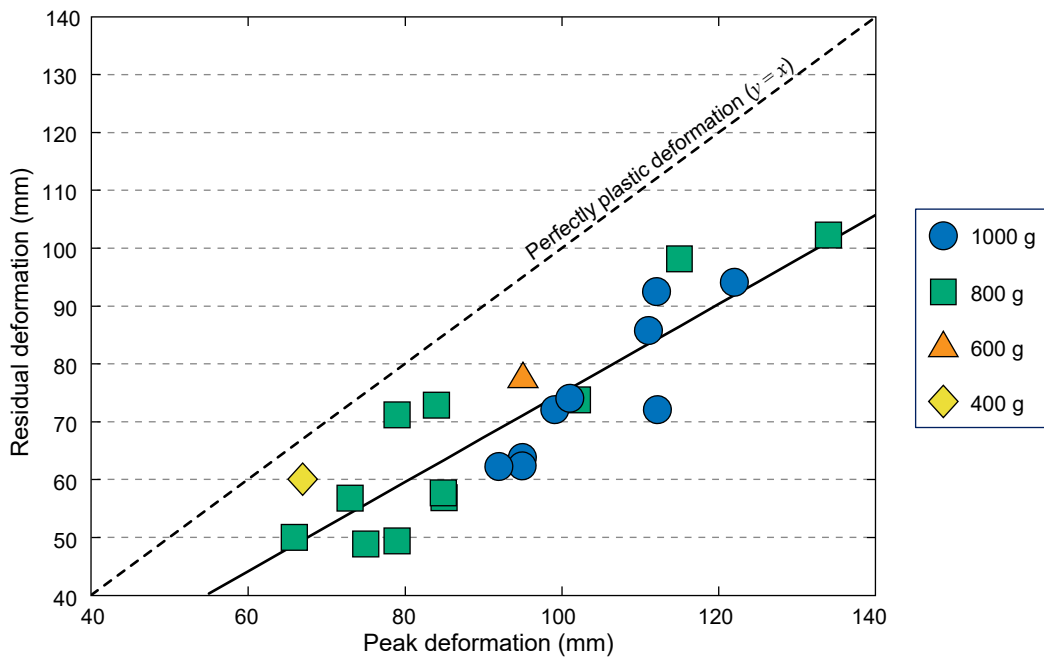


Figure 10: Residual plastic deformation of Armox 440T, as a function of peak elasto-plastic deformation for different target areal density and explosive masses, with solid black trend line

358 6. Summary and conclusions

359 Plates of Armox 440T representing half-scale models of armoured vehicle appliqué plates were
360 tested against the loading from explosive charges buried in a uniform sand (Leighton Buzzard
361 14/25). A new experimental methodology has been developed to allow repeatable testing while
362 minimising boundary effects.

363 A validation exercise was undertaken with a small number of tests using DIC. This showed
364 that the peak deformation was reached within approximately 1.5 ms of detonation before the plate

365 relaxed to its final, plastically deformed state. Overall consistency of repeat tests using the crush
366 block appeared to show a lower level of repeatability, implying that DIC is a more reliable way of
367 consistently measuring the peak deformation in the target although the calibration process is more
368 onerous which reduces the number of tests which can be conducted in a day.

369 The level of deformation measured appeared to depend on the method used to make the mea-
370 surements. Using DIC techniques provided a less variable measurement of the peak dynamic
371 deformation but produced measurements (in like-for-like tests) on average 16 mm (25%) lower. It
372 is hypothesised that this arises from momentum being imparted to the crush gauge early on in the
373 interaction which causes it to continue to compress after the deformation of the target has ceased.
374 It may be possible to minimise this effect by using a crush block which is initially stood-off from
375 the target plate to reduce the momentum imparted to the crush block. As a result care should be
376 taken when conducting testing of this type to ensure that differences in armour system performance
377 are not masked by changes in the way in which deformation is measured.

378 For the geotechnical conditions considered in this method, the peak deformation, D , as recorded
379 from an aluminium honeycomb crush block, depends on the armour areal density, A , and the det-
380 onation of a charge mass, M , according to:

$$381 \quad D = 0.24367Me^{-0.0105A} \quad (3)$$

382 The peak deformation of ArmoX 440T can therefore be used as a baseline for testing other armour
383 systems. Conducting a small number of tests of a new armour system of a given areal density
384 allows the level of deformation to be measured. Equation 3 can then be solved to determine
385 estimates of:

- 386 • the reduction in armour areal density (for a given charge mass) that the new armour can offer
387 for no increase in peak deformation; and
- 388 • the increase in threat level that can be realised without increasing the armour's areal density
389 and peak deformation.

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